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A. WILKINSON
June 1999

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Sherry's "Blue I" (July)—a quiet boat and no place contained. The camera says the picture will open—temporarily placing Stephen—somewhere with overhanging fish bones and random street-lights and bare trees.

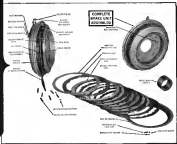
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Abstract (continued)

An important message about **LANDING GEAR** to designers of larger, faster ships



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—absorbs shock, Greater braking power without increase in brake and wheel diameter can be obtained by increasing number of discs in brake housing.

As ships increase in size and speed the need for greater braking power brings up important considerations in size and weight of wheel assembly, and in housing facilities for landing gear when retracted!

In conventional wheel assemblies where braking power is in direct ratio to brake drum area the only way to step up brake capacity is to employ a larger and heavier drum, with a corresponding increase in wheel and tire size that requires larger housing space.

Fortunately, there is a better solution. Brake capacity can now be increased to meet any load and speed requirement with relatively little growth in wheel and tire size by the use of a complete Goodyear wheel assembly—Good-

year engineered air aluminum alloy wheels, Goodyear hydraulic disc brakes and Goodyear tires, including Airbush®.

In the Goodyear assembly, braking power depends upon the number of discs in the brake housing. Brake capacity is increased by using more discs to provide larger braking area. There is no appreciable increase in diameter of either brake or wheel, hence the gain in size and weight is held to a minimum.

GOODYEAR AIRPLANE TIRE, TUBE AND WHEEL ASSEMBLY WITH HYDRAULIC DISC BRAKE
—no tire, tube and wheel shown attached on axle, easily or slide into position on the brake.

Because of this definite and easily demonstrable advantage Goodyear wheel equipment is now being specified on many of the largest ships in production today. Engineers also find that using wheels, brakes, and tires designed and built to work together as a balanced unit insures smoother, more dependable performance.

Our engineers will be glad to cooperate with you in determining the advantages of this time-proven wheel equipment in your new ships—in both conventional and triplane gear. Just write Goodyear, Aeronautics Department, Akron, Ohio, or Los Angeles, California.

(See more of The Nation's Post Office Building)

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AVIATION
June 1933



The New "105" Really Started Something!

Since 1925 we have been building fine planes which hold an international reputation for quality and safety. Yet always we planned toward the day when we could produce an equally fine plane of outstanding beauty and superior safety at a price thousands could afford.

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We believe in and expect great things of the "105". Yet we are anxious at more than \$225,000.00 worth of orders and commitments received before the "105" was publicly displayed. We are deeply grateful for this expression of confidence.

A Best of "105" Demonstration is now on foot. Dealers are rapidly being made to sell parts of the Country. We'll pay you \$1000.00 by the "105" if you plan to buy a fine airplane this year. Convince yourself that it's not about the "105" is true.

*Including Federal tax, and including price on level terrain.



The experience gained in manufacturing the Stinson 105 makes it possible to present at a single price, such a fine airplane as the "105". It is not only a fine airplane, but it is also a fine airplane. It is not only a fine airplane, but it is also a fine airplane.



PS—Remember that it will be impossible to secure deliveries of the "105" for several months and without reservation. It is true we cannot a complete reservation. It is true we cannot a complete reservation. It is true we cannot a complete reservation.

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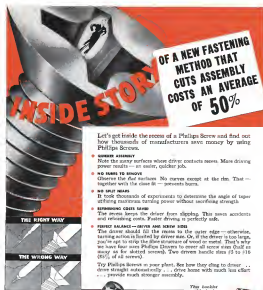
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Final report prepared

『남자친구』가 **10월 10일** 개봉한다.

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AVIATION

THE CLIVERT AMERICAN AERONAUTICAL MAGAZINE

ESTABLISHED 1910

JUNE 1939

REGULAR EDITION

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AVIATION
June, 1935
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any leads to carry forward such a program under future delivery representation. There is plenty of precedence under rulings of the courts and the Attorney General to demonstrate that the Postmaster General is not bound by the amount of current appropriations to meet the needs of the country's air mail service. Properly invoked new money and increased appropriations over old costs may be obtained, and the volume of mail increased, regardless of the size of current appropriations. It is significant that the anti-deficiency act does not specifically forbid such procedure, and it seems therefore that the CAA has a free hand to develop air or mail aviation to suit our expanding requirements. This is indeed an interesting point of view, and we hope that the CAA will recognize the implications and will act accordingly.

✱ SLOTTED WINGS as applied to Lockheed transport, discussed in an article in last month's *Aviation*, drew the following comment from Dr. G. Latham, inventor of the slot, who happened to be in this country reviewing *Handley Page, Ltd.* "I have read with great interest Mr. Hubbard's article in the May issue of *Aviation* on the permanently open slot used on the Lockheed aircraft. In fact, I had the pleasure of using this machine myself when I was in California, and he has perfectly done Mr. Hubbard about the satisfactory results obtained with this arrangement. The type of slot used by the Lockheed Company and described by Mr. Hubbard corresponds exactly with the design patented by myself in 1915 (D.R.P. 247854). This patent has expired in the meantime. The same principle was patented independently by Mr. F. Handley Page and Handley Page Ltd. in 1918. This patent has also now run in the meantime. Myers, Handley Page Ltd. flew and demonstrated the

first slotless error fixed with Gaud, i.e., permanently open slots, in 1920."

✱ MAINTENANCE will be the theme of next month's issue. Proper maintenance of aircraft on the air lines, in the services, and the private owners is one of the most important elements in flight safety. *Aviation* has long cooperated with the Maintenance Committee of the airlines and with maintenance heads of the services, and the July issue will be devoted to the work of maintenance personnel in all branches of aviation. One of the outstanding features will be an enlarged maintenance auto-book, which will present the latest and most extensive collection of maintenance data that has ever been put together anywhere. The issue will also contain the announcement of *Aviation's* Maintenance Award for the calendar year 1935.

GOODWILL AMBASSADORS in aeronautical foreign relations are Dr. George Lewis and Major Lester Gardner. As we go in press they are on the high seas on the SS "Washington", and will probably make a fairly extensive cruise in Europe before returning. Dr. Lewis is scheduled to deliver the Wilbur Wright Memorial Lecture at the annual meeting of the Royal Aeronautical Society. Major Gardner will attend the meeting to discuss the boundary between which has been avoided him. They will extend invitations to visiting officials of foreign governments to attend the Institute's International Congress of the Aeronautical Sciences scheduled for New York in September. We should all feel great admiration in knowing that aviation in America is being represented abroad by two such distinguished executives as Dr. Lewis and Major Gardner.



"Hubbard's looking his ground crew for the Elinor case."

AVIATION
June, 1935
15

Down . . . softly!



High landing speeds demand good *planning* as well as good piloting

Impact shocks may not high when the airplane land . . . a matter of simple applied mathematics. Take off and landing, run, on runway often must two smooth, oil for merely good control shock absorbing abilities in the plane's landing gear. Bendix Pneumatic Shock Struts effectively combine hydraulic and permanent cushioning, each proportionally proportional to the plane's weight and performance characteristics.

Carefully coupled with Bendix Airplane Wheels and Beams for responsive ground steering and smooth stopping, the result is a well designed, well balanced landing gear . . . something the best of pilots could for the best of performance.

BENDIX PRODUCTS DIVISION
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Bendix

LANDING-GEAR EQUIPMENT

AIRCRAFT WHEELS, BRAKES, PNEUMATIC SHOCK STRUTS, TAIL KNUCKLE ASSEMBLIES

AVIATION
June, 1950
19

Side Slips

By
ROBERT OSBORN

Bendix Notes on the N. A. C. A. Conference

Dr. Watson, former Editor of the World's Best Aeronautics Magazine (Editor), now C.A.A. Board member, on hand as usual one date with dignified manner, quiet tongue, and the remarkable watch which has, in addition to the usual works, two stop-watch bands, barometer, compass, and dials which ring out the quarter hours.

Dr. Watson, of Curtiss-Wright, who always has the largest share of sales in the latest N.A.C.A. developments, and who usually writes them up in a report when he gets back home. Each year that also successfully defends the championship for having the widest pajamas on the list.

LEWIS, completely relaxed and enjoying a trip with a crowd coming the same language. Even the women in the Chamberlain Hotel.



at Old Point didn't recognize him, not expecting him to be with a bunch of Young Persons on occasion.

As these were a number of first stars and Congressmen along, the busy and real matter, which occurred for the first time an ay

N.A.C.A. too, was generally taken to be additional holding for the San Francisco, California, literary apparatus.

HELMER and CAPLAN, the distant island situation, all in a corner together displaying the present trend toward large flying boats in the Marj.

Dr. Darrow comparing notes with Bob Korte to determine how many Teflonite would have or he could, in to get out Curtiss-Wright should considerable progress.

More to the Chief Engineer of the Norfolk and Washington Line. The Steamer "District of Columbia" is either making rough on one subject or has a tendency toward old better.

Dr. Bates did a very able job of substantiating for Dr. Ames in the morning session at the Post Theatre. However, everybody missed Dr. Ames very much and especially noted the absence of his regular greeting to the crowd at the beginning of each annual session—"Will you kindly sit down?"

ROSE CRANE looking for the fellow who gave out the statement to the newspapers that his brother couldn't have lived that percent to 500 miles per hour.

Intensive side-light on Hiram Warner. The also mentioned being played enthusiastically by the mechanical experts who know the probability against them better than the average man.

JOHN F. VICTOR, Secretary of the N.A.C.A., running around serving his usual worried look.

Center Jones hoping the weather would be hot enough on the way back that he could sell the Line a Link Traveler for his plane.

The interesting anatomy of "Desk" Lyman and "C.B." Allen frequently was altered by their shifts from newspaper work to United Aircraft and C.A.A. respectively.

Speaking of Desk Lyman recalls on that he has reported the two without a trace of his former kind, but which is not, and now for the American world. This has, which he found under a seat in the stadium after the Douglas-Tukey fight in Philadelphia, never felt the touch of breath or clothing fluid in all the years that Desk was in it, and was then kept in evidence for keeping his back (and a Pulitzer Prize). That has definitely marked the beginning of an era for America, and destined to end in



they possibly rating as a much-printed fact in the Smithsonian Institution.

FRANK RUSSELL and Warner Inland, the latter's first-hand, whose Pennsylvania forms upon the Charles Sherman Jones estate, discussing the facts and their business in the circle of life, coefficients and high-drawn tanks.

All of the sessions absolutely landing and when talk to all of the other sessions.

CHARLES WARREN DRYDEN, on the alert for new stories to add to his already remarkable collection.

Now that present and future super-velocity in the air has become such an all-important consideration in international affairs, we are wondering if the course of history might not have been changed if the countless thousands of language could have listened to the stories that were told, and the songs that were sung, by American aviation experts in the stadiums on the old District of Columbia.

AVIATION
June, 1950
17



The "Blackland," with a two-engine flight in its oval



1938 Indian champion, "BACORE," (BAC) update

A. Morris sailless type, see following pages for details



(Photos by author unless otherwise indicated.)



WINGS on the Wind

Remarkable progress in gliding and soaring in Europe is disclosed in this report on design and technique in Germany and Italy.

By Maurice A. Garbell

Member of International Research Committee for Scientific Flight

Over the Alps and across the English Channel—distance records of over 400 miles, altitude records of 30,000 ft., over 14 hours in the air on a single flight—such are the marks set by European soaring pilots in the past year. Constant research and continuous refinement in design have yielded astounding results. And the work will go on to develop new sailplanes with high speed ranges at low cruising speed, maximum in good maneuvering, combined with comfort, safety and unexcelled vision for the pilot.

Speed range is important because the sailplane needs the ability to fly slowly when circling in thermal currents and within variable clouds, and at the same time high speed is required for flying under "roads of clouds"

and for crossing down currents quickly. However, the speed range should not be ignored at the expense of the sailing speed. It has been good practice and necessity to employ wing sections with good power factor (such as the Göttinger 535) with increasing wing loading in order to obtain higher speeds. This of course has been somewhat at the expense of maximum sailing speeds and dips which were able to fly considerable distances in exceptionally favorable weather conditions were unable to use when thermal conditions were poor. As it is obviously impossible to build speed ships for such meteorological conditions, sailplane construction should have sought the same results by using wing sections with lower lift

The 10 ft. German sailplane "Solitair." Wing and nose in one piece. fuselage in two





Below: Experimenting with various airfoil shapes.



Left: A simple model in various positions on the German G-11.

Below: View from the cockpit of the German G-11.

Left: View from the cockpit of the German G-11.

Below: The cockpit of the G-11, showing the pilot's seat and controls.



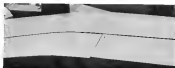
coefficients, but with better chance success at high speed and by getting a reasonable taking speed with a low wing loading (about 2.5 lb per sq ft). For example, the German builders, Jacobs, has recently been using the Gothaer 1540 section while other designers have been using such sections as the NACA 2402, 4402, 2412, 2415, 2418, 4412, 4415, 4418, etc.

The Jacobs airplane, Rotor, Wreath, and Rotor have been very successful in the last international and German events. The first two ships are of 60-hp open with moderately tapered wings. The G 545 section is used over the central 60 per cent of the span, with G 455 sections for the wing tip panels. These wings are extremely interesting because they are of a fully conventional type with a thickness factor (maximum ordinate at the wing root to span) of only 0.04. This was accomplished by using a leading edge of 100 plates to the inch for the main spar.

Rotor has a retractable tail wing while Wreath and Wreath (30 ft span) have high straight wings. Rotor also has flaps over the entire wing span to decrease the taking speed when flying in weak currents. Since its gliding angle is approximately 22 to 1, it has been necessary to provide lower taper spacers to make it possible to land in reasonably small fields. Jacobs made a reasonable compromise in favor of maneuverability by giving the wings of his ship a modulus of 7 deg between wing root



Below: Wing of German G-11 with slotted flap and aileron.

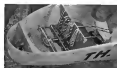


Below: Left, Wing support fitting of the German G-11.

Below: Sub-divided flap at head in wing of Jacobs Rotor.

and wing tip, moderately, the modulus is maintained also when the flaps are lowered. The degree of modulus, of course, depends not only on how much aerodynamic efficiency the designer is ready to sacrifice to obtain improvement in landing control but also upon safety considerations. Such extreme airplanes may easily reach a

speed of between 100 and 150 mph flying blind in a storm cloud and obviously such speed with lag weakness may be dangerous because of the strong negative couple set up at the wing tip. One solution to this problem has been found in the application of self-reinforcing spacers. Such spacers act not only in aerodynamic



Below: Rotor, Jacobs' machine, showing the pilot's seat and controls.



Below: Rotor, Jacobs' machine, showing the pilot's seat and controls.

Right: Two views of the Rotor, showing the pilot's seat and controls.



Below: Two views of the Rotor, showing the pilot's seat and controls.



Below: Wing of the Rotor, showing the pilot's seat and controls.



brakes to prevent the sailplane from exceeding a certain speed (for example 120 m.p.h.), but they also have a stabilizing effect keeping the plane straight below stall speed, even at low incidence without the necessity of elevator control on the part of the pilot.

The extreme conditions of temperature and humidity under which modern sailplanes fly raises a number of problems, among them the efficiency of the control. Control cables change in length under temperature differences and also an innovation in the controls themselves introduces flying difficulties. Most types of control connections compensated for temperature have reduced the difficulty in the first class, and the use of aluminum has been partially

8. German rollers with wheels of a mixed type. Left wheel shows wood and two latex bands. Wooden frame and heavy covers cover ground against wind and rain.



Many designers attempt to eliminate the old defect of all sailplanes, that is the elevator more sensitive than the ailerons and the ailerons more sensitive than the rudder. Actually, the differences are very small and the controls in a modern sailplane are practically harmonized to the so sensitivity is increased. All rollers are now strictly and dynamically balanced because the rollers themselves which sometimes occur in direct contact might otherwise induce vibration and stresses which would be destructive. Small flaps are frequently used on elevators and on ailerons to provide aerodynamic compensation.

A GREAT DEAL of thought has been put on the selection of materials for sailplanes in Europe, partly to select materials with outstanding characteristics and partly to compensate for the lack of certain materials. Most of the types are still built with skeletons of spruce or pine and covered

Typical German rollers ready for the work.

with both plywood

As mentioned previously, look forward of very thin laminates in some cases used for highly stressed wing spars. Metal wings such as those built by the Schleimann brothers in this country are almost completely a German affair. Generally, wood is the accepted wing material. In the case of the Dornier D-30 sailplane, however, we would have found that was capable of handling the stresses set up in a wing with an aspect ratio of 1 to 35 and in that case the wing spar was made of Kalsinox (a magnesium alloy) which has a specific weight of about 1.8.

Another shop, the Berlin D-6, is quite remarkable as far as design and materials are concerned. The wing is the BACA section 42012 (Schemm 1931) with the number changed from 3 per cent to 4 per cent with a slight tip of the original 23012 section. The wings are made monospar type but the flaps and ailerons have a wooden framework with Daceler covering.

The fuselage is in two parts with a main section made of steel tube with fabric covering, and with the tail made up of a framework of wooden tubes covered with Kalsinox (magnesium alloy).

CONSIDERABLE attention has been given to the design of tailfin support

members in order to relieve the tailfin drag by changing the loadings. The aerodynamic efficiency as a sailplane suffered, however, because a large part of the wing must be concentrated for stabilization and for control.

Although tailfin sailplanes have been in use for over ten years, they are thought to have significant aerodynamic qualities for thermal soaring. During the last two Rhön contests, however, this opinion has been changed by the performance of three Rhön sailplanes, whose behavior came quite close to that of several ships. One such machine has 250 miles and another one traveled at a speed of 15,000 ft. The Rhön sailplanes have very highly tapered wings with the planar root, the root and a partially movable leading edge control in the fixed control section. In some of the "Rhön" types there is a small camberline in the root, but wings are of wood or metal construction, flaps and ailerons are usually large, and few large spoilers without drag take the place of the ordinary rudder by acting as brakes on the inside wing in turns.

SENSITIVE and side movements are highly important and the three most valuable, the variometer, the air speed indicator and the two indicator have been subject to continuous improvement. Considerable confusion exists, especially about variometers, as the same instruments, sometimes and sometimes are generally assumed to be equivalent. Actually, instruments

is the indication of a very small change in vertical speed; for example, the capacity to measure a percentage from a sinking speed of 3 ft. 4 in. per second to one at 3 ft. 10 in. per second. Furthermore, on the other hand, is defined as the time required to indicate a given variation of vertical speed. For example, if the sailplane is gliding at a sinking speed of 2 ft. per second and enters an up-current which produces a sinking speed of 3 ft. per second, a variometer is "prompt" which indicates from -2 to +3 ft. per second reading in about 1 second and not in the 10 to 15 seconds usually required for rate of climb indicators on powered airplanes. Finally, precision depends upon the mechanism of variometers between the indication of the variometer and the actual vertical speed. Although a barometric variometer is generally also precise (and not very), precision and promptness are two different things which so far have not been combined successfully in the same instrument.

A considerable amount of work has been put on the two types of variometers. German types are generally based on the Schleimann membrane system which gives a maximum reliability but is not particularly prompt because of the comparatively large forces and quantities of air which are necessary to move the membrane. The two most successful "prompt" variometers have

(Turn to page 77)

Left: An interesting Indian primitive workshop for making small sail planes. Right: Indian men building small boats and sail.



A building by which shores, lower left.



Roller shown for above rollers.

solved by having several continuous dampers inside wings and fuselages. Actual bending of materials through air formation has occurred, and occasionally, a thin type of air on the wings has prevented movements of ailerons.



An Indian workshop making small sail planes. Right: Indian men building small boats and sail.

Indian type which, besides of little loss, may be described as a 70.

Pilots by the

THOUSANDS

By Oliver L. Parks
President, Parks Air College, Inc.



There is a remarkably efficient and dependable service that the airlines have been giving, together with the Spanish Civil War and subsequent events in Europe, have focused public attention on aviation as never before, and most of this attention is favorable. The significance of this modern method of transportation, both in our peace-time economy and in defense, is rapidly penetrating the national consciousness.

An immediate and tremendously important result of the greatly accelerated widespread interest in aviation is that our administration leaders and Congress have proposed and secured favorable action on plans for the acceleration of aviation's development. The plans which must directly affect those of us who have devoted our energies and resources for a number of years to the development of aviation schools and centers for those both affluent and unaffluent aspirants, have to do with the training of aviation personnel.

The question immediately arises in our minds as to what will be the portion of our efforts in the rapid march of personnel training needs. Will they be in the training of leadership? Will they be somewhere back in the process? Or will they be slanted out of the line of march altogether?

Taking for granted the ability and desire of those in authority to avoid mistakes, and, in order to avoid waste, to utilize as fully as possible existing physical and human resources, we can conclude that the answer depends, in part at least, upon the ability of the schools themselves to provide training of the character and in the volume required.

The prospect of doubling the existing number of pilots within a year has created many training problems. Steps are now being taken by the C.A.A. to provide solutions for those now foreseen, and for those which will be revealed in the course of the experimental program now under way.

We asked Mr. Parks recently to comment on the situation as he saw it. His views are particularly valuable since his experience includes operation of a large residence school, as well as a smaller one of the type now being used in the C.A.A. Civilian Training Program.

As is well understood, at best within the industry, long plans for personnel training are underway or are proposed. The fact, the Civil Aeronautics Authority plan to provide flight training up to the Private Pilot stage for 300 selected applicants in 13 designated colleges and universities, is paving the highway much at the time due is within. By the latter half of June it should be completed, this interim rule, and the lesson to be learned from this trial effort available as guide for future planning.

At the present time, however, I can visualize the opinion that one of the lessons that will be emphasized is that the success of any training venture is directly dependent upon the caliber of the training talent. It will be emphasized that in aviation, as in any other field, skill in teaching is required. Just as a famous athlete may be a failure as a coach, so a capable pilot may not be altogether successful as an instructor. Class structure, active guidance and military flying are all valuable, but so any background must be added the necessary patience and nurturing technique if the flight instructor is to be satisfactory work.

Another lesson that this preliminary training experience can be expected to teach is that sound business principles can be observed in advantage in a pilot training program as elsewhere. For example, we must and must have capable instructors, and we should require to pay for their services a sum adequate to maintain their interest in their work and to hold them. If we don't have to pay it, we should pay a anyhow just to avoid demoralizing and lowering down the whole program level on. And sound business principles dictate that salaries be paid out of income and not out of capital.

For there are other times to be paid out of income. It must mean the law requires—and it all states good business principles require—that work-

men's compensation, expenses be covered. There are also other forms of insurance to be carried, equipment to be maintained, depreciation to be written off, taxes to be paid, utilities to be taken care of, a reserve-for-contingencies fund to be set aside and a means to be made an internal capital, all out of income. To operate otherwise is to operate on a shoestring and to realize sound business principles, and we can all agree that the "shoestring" era in aviation has continued long enough.

When all the trends of this initial program are so, I am convinced an important recommendation of them will show that a minimum of \$200 for a minimum of 30 hours primary flight training is not enough to permit the instructor to observe sound business principles. Since in aviation it is never wise to purchase quality at the expense of quality, the answer is adequate pay, say \$500 or more, even though the total hours of training is less.

Whatever the lesson learned, they should be decidedly helpful in charting the course of the second plan, that all pending training for as many as 60,000 or even nearly 100,000 young men annually. Obviously present flying practices and methods, high standards will need to be maintained if for no other reason than that the law of gravity is still operative and a series of accidents would surely be such to destroy the public acceptance of aviation that has so patiently been built up. The maintenance of standards is a comparatively simple matter provided the flight instructor's attitude is an established automatic recognition.

It is here the flight schools which have been developed by the Authority can cooperate effectively by giving to instructors who would propagate a course of progressive training that has been outlined in complete detail by the Authority. The success of the entire program would be

further assured if each instructor were given three weeks additional training by the Army Air Corps at Randolph Field. Such an objective is to create a reservoir from which pilots would be drawn readily in case of national emergency, it is possible for personnel to be made for the military training.

Just what part, if any, commercial aviation schools will be asked to play in the third and fourth plans, that of greatly increasing the flight and mechanics personnel of the Army Air Corps, I do not know. Probably details will be revealed before this is printed. Without doubt, any school called upon will gladly cooperate to the fullest extent possible, making available its resources as may be required.

Each cooperation must stand in of value to the Army Air Corps—otherwise, there would be no reason for entering into it—and in two ways it may be of value to the schools. The first way and the most obvious is the financial. If the payments for training given are sufficiently large, the school can improve its equipment and strengthen its financial position. The second way is of equally great or greater importance. Through association with officials of the Air Corps and familiarization with standards maintained and the training methods used, the school can improve the experience already gained and so increasingly improve its training service.

I am finally convinced, however, that it would be a mistake for any school to look at participation in the Air Corps program as an end in itself. I realize this seems very clear but it is not. The point I mean is no airplane is that in preparing pilots for all sorts of careers in commercial aviation, the commercial aviation schools of job security of their best efforts and, because the industry has possibilities for almost limitless expansion, one that will keep them occupied for many years.

In all the confusion of events, reports and counter-reports, I am convinced there has appeared no means by which aviation schools here or there a constructive educational job should also their long time objective. Cooperation in carrying on. Federally sponsored plans for personnel training may require considerable readjustment in the administration of school affairs, but instead of interrupting, this cooperation should reinforce the school's program for improvement and development.



Airplane Type	Avg. Size		Avg. Weight		Avg. Price		Avg. Price		Avg. Price		Avg. Price	
	sq. ft.	lb.	sq. ft.	lb.	\$	%	\$	%	\$	%	\$	%
Army	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000
Naval	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000
Marine	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000
Army	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000
Naval	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000
Marine	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000	100	10,000

* Includes engine and propeller. † Excludes engine and propeller.

the capacity of the industry should be somewhat in excess of requirements when working under ideal conditions. "Ordinary" design margins, without service test, is at least a 10% margin and it is to be hoped that movement to such method of procurement can be held to an absolute minimum.

For determining basic production capability, I have selected three factors, one building airplanes, one engines, and the other propellers, with which I am intimately familiar. The airplane factory is producing planes on orders which, in contrast, approximate the program scheduled in the next two years. These factories are all working now on close to a week's basis. In any such factory there will be a certain amount of equipment to maintain production at a rate which is to equip a plant from the machine tool standpoint as to operate into an air shift in an efficient manner. There is, occasional airplanes, even as a one-shift factory, is continued to never-ending operations which may have its roots down from general production plants.

Table 1 gives the basic data on actual production of a plant when on an air shift and also when extended to capacity. Under capacity conditions,

a one per cent time spare expansion is assumed to provide labor balance. A labor commission per square foot which cannot be exceeded efficiently by more shifts than those indicated for the capacity contract, is derived. To the cost of engine and propeller, greater resource in subcontracting is preferable with smaller provision for providing the number of square feet required per man. Data on unit cost per point appears in Table 1. This item for unit cost is being the most readily determinable from the type of information available on the various classes of aircraft.

Engine commission per square foot is based on engine of "average" which is gross weight minus the sum of useful load plus weight of engine, propeller and instruments. The "average" engine being manufactured as the particular factory studied differs 15,000 lbs. (average) horsepower, weighs 12 pounds per horsepower and can be purchased approximately \$100 per horsepower, or \$250 per pound. The weight shows that propellers will approximate engine in cost per point so that in subsequent investigations it is permissible to group these two data. Production in terms of dollars per square foot per year are given to compare with recent estimates when

such criteria. This method is not believed to be quite as satisfactory, however, as the intra-perman basis and here.

It is also important to note that in such case, a balanced condition of quantity production and of replacement work was under way in the plants studied. To the cost of airplanes, sixty-five per cent of the direct labor force was engaged in producing a type currently being delivered thirty per cent of the total of the total production order, and two per cent an experimental work. The general efficiency of the plants studied for obtaining the maximum production is believed to be at least equal to the average of the industry later during the expansion and therefore suitable to provide such basic data.

Table 2 provides information for the aircraft industry in the country from the standpoint of floor area, labor force, and square feet of floor area per horsepower for the three main divisions of the industry. The study shows that propellers will approximate engine in cost per point so that in subsequent investigations it is permissible to group these two data. Production in terms of dollars per square foot per year are given to compare with recent estimates when

Table 1
Estimate of Square and Total "Equivalent" Airplanes and Engines
Airplane

Airplane	Production	Price	Equivalent	Total
Army	100	10,000	100	10,000
Naval	100	10,000	100	10,000
Marine	100	10,000	100	10,000
Army	100	10,000	100	10,000
Naval	100	10,000	100	10,000
Marine	100	10,000	100	10,000

* Based on "Equivalent" airplane as produced by July 1, 1941.

Engines

Engine	Production	Price	Equivalent	Total
Army	100	10,000	100	10,000
Naval	100	10,000	100	10,000
Marine	100	10,000	100	10,000
Army	100	10,000	100	10,000
Naval	100	10,000	100	10,000
Marine	100	10,000	100	10,000

Assumed Composition of U. S. Army and Navy Air Forces

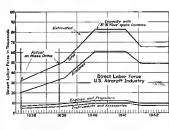
Aircraft	Percentage	Engine	Percentage
Fighter - 10 types	10	Engine	10
Bomber - 10 types	10	Engine	10
Engine - 10 types	10	Engine	10
Engine - 10 types	10	Engine	10
Engine - 10 types	10	Engine	10
Engine - 10 types	10	Engine	10

Note: Figures have been derived from published record of Congressional Budget and Appropriations Committee.

can be fulfilled. There then appears a situation, including engine and transport plants (which are maintained by many of the same firms of the industry who produce military equipment for the Government) where the holding up of quantities to give the required total, allowing for expansion which will inevitably occur when the expansion program is under running, it should be noted that in meeting a rapidly expanding program, the rate of absorption will be less than will later hold when maintaining the force after the program is realized. At this point, emphasis should be placed on the great economy for allowing adequacy for the production of more engine and transport orders so may be available or forthcoming later, so the maintenance of these plants is necessary important to the welfare of the country as well as of the industry, particularly when looking forward to the time after our own military expansion program is accomplished. Measuring these markets currently in the only way that can be used is after the fact.

In Table 3 the effect of years is added to the airplane program as shown in Table 4, to give total equivalent airplanes and engines. The net result is that between now and July 1, 1941, three million to be produced 11,000 airplanes and 14,000 engine propellers units. Actually, because of single-engine planes being built, there will be need for a greater number of engine units of smaller weight and power. However, this comparison to be produced will work out correctly when using the "average" "larger" engine.

In order accurately to estimate our capacity, we must know just what we are going to build, what is the program, and what is the program. For this purpose, it is necessary to know the approximate composition of our air force by type. Such an estimate appears in Table 5 which, it is believed, is self-explanatory except to note that under each general classification, defined by function, there are respectively two or three types of airplane each, for instance, to order "Pursuit" where there are single-engine single-place, single-engine two-place, and two-engine types. It should



be noted that in these estimates, all airplanes, no matter what function they perform, are under consideration the picture as being confined by reference to "single" type. "First line," or "main" type, which are under development a comparative analysis of the air force has developed plans continuing with the war effort in other fields, one maintained or renewed. Naturally, these curves will not be straight lines as shown, simply changing at given points, but will be rounded off to take smooth curves. Again it should be mentioned that consistently more subcontracting outside of the aircraft industry will be anticipated in the construction of engine and propeller units than will exist for the airplane shell. These curves present establishing an average labor force for the first year of the expansion the second year being considered as the capacity figure previously worked out.

By the use of the data obtained in the foregoing tables, it is possible to develop Table 9 and 10 which show, respectively, the airplane and the engine, the performance capacity of the industry. Having with known accomplishment, figures, naturally "weighted" for area of order and type of construction, as is shown in

(Page in page 12)

AC FOR AIRCRAFT



WHILE some of the early types of airplanes carried a storage battery, it was inevitably used for ignition and, therefore, cannot be considered as intended for supplying auxiliary electric power. The battery, however, was used for this purpose when instrument board lights—the first demand for auxiliary electric power—were installed. Later came night flying with navigation and landing lights, then radio receivers and transmitters, each increasing the power demand. The promise of retracting landing gear brought with it the first application of electric motors on aircraft. Increases in passenger comfort introduced the demand for cabin and heating lights. Higher speed planes accumulated more powerful navigation lights, while more extensive installations demanded more radio sets.

The increase in auxiliary electric power demand is shown in Figure 1. It was realized in 1934 that the storage battery, even though it was charged by a 30-ampere generator, was no longer capable of meeting the demand placed on it and an investigation of some other source of possible electric power was begun. Figure 1 does not show the demand for auxiliary power in the form of parasitic power for heaters and pressurization and hydraulic power for auto-pilot and

The one hundred and one gadgets that have been added to the modern airplane have boosted demands for electrical power to the point where auxiliary generator sets are coming in to the picture. We asked

P. C. Sandretto

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to round up the present situation regarding power supply, and point up some of the problems for the immediate future.

brakes, which came with the advent of the modern airplane. To meet this demand, vacuum and hydraulic gauges were necessary and increased the number of power take-offs from the main engine to the point where maintenance personnel believed that these power take-offs would affect the reliability of the main engine. Since the removal of two aircraft instruments would save 712 lbs., it was thought that it might be possible to justify a separate generator engine which would drive a generator to supply electric power and also drive all other accessories which had previously been attached to the main engine. Since this step would violate the post-war doctrine, it was decided that a thorough study

of all types of electric power should be made in order to determine which one was best suited to aircraft application. It was realized that the airplane engine would not involve transmission, lights, and motor accessories, which include the generating and distributing of electric power as well. The study was therefore made on this basis. One of the purposes of the paper is to present these findings.

The first decision was whether the new source of electric power was to supply alternating or direct current. Alternating current has the following two advantages: 1.—It can readily be transformed to any voltage desired with the use of rotating machinery. 2.—Alternating current motors do not

require brushes and are therefore simpler to maintain since no replacing brushes, they do not generate noise in the motor system. Direct current has the following two advantages: 1.—It can be stored and, hence, be available without operating the primary power source. 2.—Direct current motors have better torque characteristics than alternating current motors and are somewhat lighter for a given shaft speed and horsepower. Because of the number of different voltages required for radio purposes, the first advantage given for alternating current was believed to be very important; the second advantage to be of some value.

It was decided that for communications systems, the ability to store direct current was of little importance since landing in emergency fields is a very serious occurrence. It was believed that the second advantage for direct current was not a major factor, so the decision was made in favor of A. C.

No great consideration was given to a standard A. C./D. C. source because it was thought necessary to pre-determine the amounts of each kind of power required for each application and provide generating equipment for both kinds of power. Elaborating, a night be desirable to use strictly all

direct current power during one portion of a flight and strictly all alternating power during another portion, so the generating equipment for an A. C./D. C. system would have to be twice as great as for an A. C. only or D. C. only system.

Today we can write of the various discussions concerning the problem of arriving at a standard of A. C. power for aircraft and can give a summary of the points made by numerous engineers who have taken part in these discussions, but in the fall of 1934 when the writer first began his study of this subject, there was little information available and it was difficult to find an transport engineer who would take any but a casual interest in the problem. True, some of the Service had utilized A. C. power on aircraft, but their application had been limited to radio plate supply. In 1933 United Air Lines developed a radio receiver which could be adapted to A. C. operation, but the investigation of A. C. as applied to large aircraft was not begun until 1934. In January of 1935 a study of A. C. up to 200 volts, by Hertz and Kaufman of San Francisco, was distributed among engineers of various aircraft companies. In June of the same year, the Eclipse Aircraft Corporation published an A. C. power study.

This publication was restricted in its distribution. In June of 1938, Sub-committee #2 of the Radio Technical Committee for Aeronautics was appointed by the Bureau of Air Commerce for the purpose of studying A. C. power for aircraft. Without attempting to go into detail covering the points made by the various studies, the writer will attempt to sum up the substance of the various discussions.

Voltages

Following the decision that the power supplied would be A. C., work was done to determine the optimum voltages of this new supply. Complete electrical systems were designed for an airplane about the size of the Sikorsky 5-40 in the investigation that the smallest size which had enough mechanical strength to allow its being pulled through cockpit without damage was 250 A. W. O. Two designs gave the data plotted in Figure 2. With the wiring weight of a 12 volt installation calculated as being 180%, the weight of a 120-volt system would be 30% and a 220-volt system 8%. Better than the curve does show, it appears generally need a voltage of 300 is met, but that the weight advantage between a 120 and a 220 volt system (Times to page 47)



Stacks and Rings

Using the exhaust gas for heating carburetor air and displacing carbon

Part III—Carburetor Air Preheat

By George F. Titterton

Aviation Engineer, Grumman Aircraft Engineering Corp.

The temperature of the intake air entering in a carburetor averages 7° lower than the outside air temperature. This drop in temperature is due to the cooling effect of the fuel in vaporizing and the expansion of the mixture. It is shown that on a climb dip with the air temperature between 30° F. and 50° F., ideal conditions exist for the formation of ice in the carburetor. This condition can be eliminated by heating hot air to the carburetor. Various opinions exist as to how much the air temperature must be increased to be satisfactory under all conditions. The carburetor temperature rise over outside air temperature is believed to be

equal to a three-degree rise in dry bulb air temperature. (See Table I, page 10.)

60° F. This temperature rise is impossible for the dip in the outside temperature. At times when the air temperature is below 40° F., the air is relatively dry and there is little moisture for the cold mixture to freeze. As long as temperatures above 40° F. the 60° F. rise in hot air will be sufficient to keep the carburetor walls above the freezing point.

The device shown in Fig. 1 of the first article is capable of raising the air temperature above 90° F. This temperature increase is sufficient to provide a large margin for all weather conditions. A number shown is shown in Fig. 12. A collector with a 3-inch radiator tube can raise the air temperature somewhat over 60° F. The temperature rise should be main-

tained when the engine is developing 20-hp. of its normal rated power. The power approximately the light condition of which the engine would be started under some conditions. It is common practice to fly airplanes at all times using air preheated to 90° F., to prevent any possibility of ice forming in the carburetor.

Types

There are many types of exhaust collectors and they may be classified in several ways. We have the shroud or wall type, the insulator type, and the hot spot type. In addition we have walls for heating carbon, and exhaust driven superchargers.

Shroud Type Exhaust Collector

The exhaust collectors shown in Figs. 1 and 12 are typical examples of a shroud type collector. When the side cooling ponds are heated in



Fig. 11. Exhaust manifold for Wright Cyclone R-1820-23 engine. Side long ducts exhaust separately with main ducts side protecting from each end. Also in-between ducts is main collector ring.



Fig. 12. Duct assembly for Wright R-1820-23 engine. Side down shroud made to match engine heated liquid-coolers. These liquid-coolers absorb movement between engine which are mounted rigidly in the engine structure and the main collector ring mounted in engine.



Fig. 13. Exhaust manifold for P & W R-1500 engine. Side wall for carburetor hot spot and wall around main collector ring for heating carbon air.



Fig. 14. Exhaust manifold for P & W R-1500 engine. Side complete barrel wall for pre-heating carburetor air.



Fig. 15. Cooling shroud and practical exhaust manifold for P & W R-1500 engine. The heated air from within the shroud is led to the carburetor air screen.



Fig. 16. Exhaust manifold for Wright Cyclone R-1820-23 engine. Side wall around main collector for pre-heating carburetor air with connection to carburetor screen. Side also the exhaust connections for the carburetor to provide heat for the inside screen.

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place the upper two-thirds of the carburetor collector is completely enclosed. At the upper end the cooling enclosure or shroud draws into an entrance in the carburetor air scoop. This air scoop entrance is valve-controlled so that the hot air rising from the shroud may be either spilled into the atmosphere when not needed, or drawn into the carburetor. The spilling of the hot air provides ventilation and prevents overheating and burning

of the shroud trailing. The hot air is drawn into the carburetor to prevent icing as explained under the sub-heading "Carburetor Air Preheat." The wall shown in Fig. 15 works on the same principle as the shroud. It is attached directly to the collector and is independent of the cooling. The wall is made of the same material as the shroud. The cooling thus forms the shroud may be continuous or broken short, or the

ordinary aluminum alloy cooling material if it does not touch the collector.

The flow of air through the shroud is caused almost entirely by the action of the carburetor. Frequently this flow is not rapid enough to supply the required amount of air and irregular engine operation results. To overcome this difficulty it is advisable to install an air scoop extending

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Snatching the Mail

All-American Aviation Starts
Airplane Pick-up Service

When our reporters gathered down by the railroad tracks to watch the train continued day pick-off at 40 m.p.h., they were treated to a picturesque spectacle. Now we ran down to the rail support, as never to handle, and as the airplane came down between a pair of sleepers and caught off a mail pouch at 100 m.p.h. If we watch closely we can see delivery men make short touchdowns. Regular mail service was inaugurated May 12 by All-American Aviation, Inc., on two routes—No. 302, Philadelphia to Pittsburgh and No. 100, Pittsburgh, Washington, Charlotte. Intermediate stops will be added on a staggered schedule and, by early July, 50 cities will be served, as indicated by a year's experimental contract recently awarded by the Post Office Department.

Mail Service (M)

Mail Route	City	Mail Service (M)	Starting Date
1	Philadelphia, Pa.	May 12	May 12
2	Pittsburgh, Pa.	May 12	May 12
3	Washington, D.C.	May 12	May 12
4	Charlotte, N.C.	May 12	May 12
5	Philadelphia, Pa.	May 12	May 12
6	Pittsburgh, Pa.	May 12	May 12
7	Washington, D.C.	May 12	May 12
8	Charlotte, N.C.	May 12	May 12
9	Philadelphia, Pa.	May 12	May 12
10	Pittsburgh, Pa.	May 12	May 12
11	Washington, D.C.	May 12	May 12
12	Charlotte, N.C.	May 12	May 12
13	Philadelphia, Pa.	May 12	May 12
14	Pittsburgh, Pa.	May 12	May 12
15	Washington, D.C.	May 12	May 12
16	Charlotte, N.C.	May 12	May 12
17	Philadelphia, Pa.	May 12	May 12
18	Pittsburgh, Pa.	May 12	May 12
19	Washington, D.C.	May 12	May 12
20	Charlotte, N.C.	May 12	May 12



Operations Manager James Ray has been busy holding up his signature, one which now includes Phil, Holger Harris, Thomas F. Klinefelter, James V. Parnell, Norman Bennett.

Mail Service (M)

Mail Route	City	Mail Service (M)	Starting Date
1	Philadelphia, Pa.	May 12	May 12
2	Pittsburgh, Pa.	May 12	May 12
3	Washington, D.C.	May 12	May 12
4	Charlotte, N.C.	May 12	May 12
5	Philadelphia, Pa.	May 12	May 12
6	Pittsburgh, Pa.	May 12	May 12
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17	Philadelphia, Pa.	May 12	May 12
18	Pittsburgh, Pa.	May 12	May 12
19	Washington, D.C.	May 12	May 12
20	Charlotte, N.C.	May 12	May 12



Charles D. Vane, and Floyd J. Julian. Second member of the two-man crew is called tentatively the "pickup man." His job is to examine the mail when he finds the sleeping bag and load up the mailroom and at each stop, and see pouches between stops. Pioneer sky mail clerks so far selected include (From left to right):



- A-Mail box loaded mail inside the plane and we used to station pouches
- B-A bag ready to be dropped through the hole in the cabin floor. The cable release is at the center and the latch at the right
- C-Detail of the latch with the line slack stretched at the right
- D-Mail bag suspended and ready for pick up
- E-Detail of hook attachment of release

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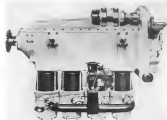
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Monsoon 100 hp. Engine

American Version of French Regnier box Hehson Carburetor



DETAILS. Offices of the company are Joseph L. Cobb, president and director, John J. Starnes, vice president and director, Walter E. Hume, director, Robert E. Hume, director, and J. Eusebio Hume, secretary, treasurer, and director.

With a dry weight of but 365 lbs., the Monsoon is said to be the lightest engine in the world in its power class. Design is conventional, with four cylinders arranged in-line, inverted and air-cooled. Features include fully enclosed valve gear, overhead gas cylinder valves, and a triple oiling system that provides positive lubrication in all positions, including operation

down flight. The engine is being produced at a low price, and it is also claimed that operation and maintenance costs are unusually low. Fuel consumption is given as 42 lb./hp hr., and oil consumption, 406 3/4 lb./hr.

Construction provides a heat-treated cast aluminum alloy crankcase, with magnesium alloy cover and stationary case. Four detachable mounting arms bolt to the sides of the case. The four-throw forged steel crankshaft is mounted in five main bearings. Connecting rods are of forged duralumin and pistons of heat-treated cast aluminum alloy with two compression and one oil ring. Cylinder assemblies consist of a forged steel barrel, machined all over and fitted, with a heat-treated aluminum alloy head bolted to the cylinder end and bolted to the magnesium and cast iron base. Magnesium and the accessory drive are mounted at the top rear of the engine.

Specifications by the manufacturer:
Type: Light, single, inverted, air-cooled, 100 hp.
Cylinders: 4
Stroke: 4 in.
Bore: 4 in.
Horsepower: 100 at 2800 rpm at sea level
Crankshaft: 4 in. dia.
Oil consumption: 406 3/4 lb./hr.
Fuel consumption: 42 lb./hp hr.
Weight: 365 lb.
Dry weight: 365 lb.
Fuel pump: 100 lb./hr.
Accessories: Magnesium alloy (Aluminum alloy) carburetor, fuel pump (magnesium), 100 lb./hr. pump.

Why is this strut Aluminum?



(Mark what you think is the best reason, then read below for the designer's reason.)

- ☐ To get lightness
- ☐ To assure soundness
- ☐ To save money

THREE ALUMINUM ALLOY strut designers would concur: a typical 5000 Series aluminum strut is greatly preferable.

It was chosen because it saves weight. It was indeed selected because it will remain sound and strong. But another important factor which made the manufacturer specify Aluminum was this: In a judicious selection of the most suitable Alcoa Aluminum Alloy, the strut costs less than a strut made of other material considered adequate.



There are two significant facts behind this comment. One is that the wall thickness of the strut must be great enough to be stiff and prevent buckling. It follows, since Aluminum is light, that fewer pounds of Aluminum were required for the desired thickness. Metal is sold by weight.

so lower weight brought relatively lower cost.

The second fact making for economy is that production of Alcoa Aluminum Alloys is set up for aircraft requirements. Efficient sections could be selected, less than for which drawing tools are available, without need change to the customer. The necessary inspection is routine for alloys of Alcoa Aluminum used in aircraft. Hence extraordinary costs are not involved.

It has taken this strut one more case demonstrating two important points:

Alcoa Aluminum Alloys are ideally suited to the engineering requirements of aircraft. And the production of Alcoa Aluminum Alloys is planned to fit the production requirements of aircraft manufacturers. ALCOA Aluminum Company of America, 3302 Gulf Building, Pittsburgh, Pennsylvania.

ALCOA ALUMINUM



"Hush-Hush" Boat

Twine engined Consolidated combines high speed and long range

Consolidated's "hush-hush" seaplane, the Model 31, has captured flying boat, took to the air May 18th with William Whistler at the controls. The new ship, a high-wing cantilever monoplane, of all metal construction, combines high speed, long range, and heavy-carrying power in a plane of relatively small size. It is powered with two Wright 18-cylinder turbo-prop engines rated 2,000 hp. for takeoff, driving 140 dia. Hamilton-Standard 3 blade propellers.

Stability on the water is provided by outrigger floats which mount against the under side of the wing in flight. The relatively high taper of the wing provides by extension of the floats in the FEV series. A bi-cycle beaching gear is permanently attached and is also fully retractable within hull walls. Although intended strictly as a beaching gear, it is believed rugged enough to withstand airport landings in an emergency.

Gross weight is approximately 50,

800 lb., wing span 118 ft., overall length 73 ft., height 32 ft., and maximum wing chord 34 ft. Although little detailed information is available, it is understood to have a wing with high taper ratio, an aspect ratio of approximately ten to one, and a maximum loading of more than 15 lb. per sq. ft. Gross wing area is reported at approximately 1,000 sq. ft. Of particular interest is the use of a new type airtail recently developed by Donald H. Davis, independent aeronautical engineer.

The Davis airtail is reported to possess a number of interesting characteristics, including low drag and low center of pressure (tail). Hydrodynamically optimized Fairley wing shape assist landings and takeoffs. Although no performance estimates have been released, engineers have estimated the plane to have a top speed ranging from 255 mph to upward at 300 mph, and an all-out maximum range of 4,000 to 5,000 miles.

Reports of preliminary testing indicate that the outboard floats which mount against the under side of the wing in flight, provide excellent stability on the water, quick take-off, and smooth landing. The hull is of welded design and is said to be the result of the cantilever wing technique in the Langley Field testing house, where complete scale models are now being "stressed" off the water instead of building test runs with a bare hull. The Model 31 differs materially from current hull designs, having unusual depth of hull combined with a relatively narrow planing bottom, and a positively stable appearance.

In comparison the Consolidated Model 31 is of conventional design. The wing, tail section, and planing surface of the hull bottom are all thick covered. All doors, flaps, radars and elevators are fabric covered. The tail consists of a conventional stabilizer with twin fins and radars mounted at the ends.

Two flight decks are provided in the hull, one above the other, providing a large flight control compartment, and eight additional compartments for passengers and payload. As a commercial vehicle the Model 31 has a day capacity of 32 passengers and a night capacity of 26 passengers, with sleeping accommodations. It is said to be capable of operating in trans-Atlantic service with a load of 20 passengers. Cruising speed on 30 gpm power has been unofficially estimated at better than 200 mph. As a military airplane the Model 31 would support the FEV line of two-engine patrol bombers which it greatly exceeds in speed, range, and load-carrying ability. However, the plane is a private venture on the part of Consolidated Aircraft Corp., not having been subject to the military procurement process which adds so severely to the cost and time for construction of prototype military aircraft. As a result the design was developed, built and test flown within a period of only five months, instead of the two to three years needed for similar under military procurement procedure.

Military men who have knowledge of the capabilities of the Model 31 have their high estimate of its naval effectiveness primarily on its ability to locate and destroy with relative ease all but the most heavily armed surface vessels. With its high speed, long range, and ability to carry a heavy bomb load, the Model 31 is believed capable of engaging large areas of ocean entirely free of all enemy vessels except heavily protected fleets of heavy warships. It is pointed out that its high speed makes the plane nearly immune against attack by current planes of current design, and necessarily reduces effectiveness of weapons for the same reason. Yet the beach carrying capacity is sufficient to cripple any ship when it hits but once by a bomb of the maximum size this plane can carry.

No further information on the plane is available as we go to press, but aviation people everywhere may well watch with interest the word of civil or military application of the Consolidated Model 31.



A distinctive feature of the Model 31 is the permanent beaching gear of the bi-cycle type which is fully retractable within walls of the hull. Although it is believed extremely rugged, the gear is believed to be rugged enough to withstand airport landings in an emergency.

Snatching the Mail

(Continued from page 35)

Glynn Byrnes, DeHon R. Osborne, and Victor Treutwein, Headquarters of the firm of von Bremen, Baltimore, is under the direction of Charles H. Cooke, as the Pittsburgh headquarters. John R. Easley is a contract operations manager.

Mechanical details of the pick-up device have been worked out over many months of experimenting by Vice-President E. J. Adams and President Richard C. de Vries. In the process, the operation is extremely simple. On the ground all that is required is two steel wires, 50 ft. long with one foot sections spaced them to make them more easily distinguishable from the air. Each unit is fitted with an ordinary pulley and rope with a weight and slider hook designed to open at about 125-150 ft. pull. The crew rope, on which the bag is free to travel, is attached to the handle on

either end and rated into position by a ground crew at one or two times. The bag drops, a slide fitted with a strong grapple which engages the conveyor on either side of the bag. The grapple slides in a least in one end of the rope while the bag slides to the other end and the pick-up crew pulls on both rope and cable by a winch mounted on a long shock absorber in the coils of the ship. The bag has delivery it suspended on a 60 ft. cable and released by the pilot before the pick-up contact is made. With a link greater it is possible to land the delivered bag within a few feet of the poles. Bags are provided by a rough, light-weight material to protect bag contents. Landing facilities are necessary on very fine open is required for the pick-up device and no provision for landings is made in the schedule.



Clump of the Consolidated Model 31 showing the method of retracting the wing floats and the permanent retractable beaching gear.



From left, Byrnes and Victor Treutwein and DeHon R. Osborne (left to right) discuss the installation of the experimental Model 31 at Consolidated, Pa.

Greenwood-Yates
Monoplane

Shedding in the Northwest: From Portland, Ore., comes news of the Greenwood Urban Monastery, a self-sustained city with both wine and houses of quality construction. For further details see page 60.



Eagle of Iran

New Tynian Cured his Sea Scurvy by Recourse to this Superior Remedy designed especially for the late King and built in his Majesty's apothecary. Several testimonials from the most and famous Sea-Officers and Captains are. A plentiful supply. One Bottle of being in possession with a Glass & Whetstone, White Amber, Venice and London Bitters, Orange-Tincture, Peppermint-Oil, and a few Drops of each, will cure and prevent the Scurvy in every Active Person, and Effectually remove the Venereal and other venereal Disorders. Glass apothecary jars used include Glass Bitter-Glasses, Withen-Concoction, and many special Instruments.



Lycoming 65

For the latest information on the Lymington line just introduced, see Model 85, diesel drive, horizontally opposed type available with single or dual engine options. Designated OGG85 Series, the new engine is approved by the C. A. R. for a rating of 75 hp. at 2300 r.p.m. and will be increased on the "EE" model over 75 hp. to 85 hp.



**CONSOLIDATED LONG-RANGE
52 PASSENGER FLYING BOAT**
powered by
TWO DUPLEX-CYCLONES

Cessna Aircraft Corporation's new long-range, 53-passenger Flying Boat introduced the 18-cylinder 2000 H.P. Wright Duplex-Cyclone aircraft engine.

As a commercial airliner, the Consolidated Model 31 has a maximum capacity of 32-passengers and a crew of five for day operations and sleeping accommodations for 20 passengers for protracted flights or trans-oceanic service.

Although no performance figures may be missed, the total of 4000 H.P. delivered by the two Duplex-Cyclones has given the engine an exceptionally high speed, range and load capacity.

Outstanding features of the Model 31 include two ball locks—one above the other—providing a large tight sealed compartment and eight additional large compartments.

The ship is equipped with a hydraulically operated, retractable tricycle beaching gear—the first application of such equipment.

"Fly Fish Wright The World Over"

WRIGHT AERONAUTICAL CORPORATION
Phoenix New Jersey
A Division of Curtiss-Wright Corporation



WRIGHT *Aircraft* ENGINES

that CRUCIAL moment

... when a 44-pound engine mount pulled a 1200-pound whirling mass safely out of a 600 M.P.H. power dive.

That moment was a real test and proved the strength of **OHIO SPECIAL QUALITY SEAMLESS TUBING**.

Only a pilot can fully appreciate the raw razzing and value of carefully chromium-plated stainless steel tip by stress and power used at the end of a dive. When that crucial moment comes, as in the recent record set by Test Pilot H. Lloyd Chalk in the Curtiss Hawk 75-A Pursuit Plane, pilots take their lives on the ability of materials to stand the stresses of tension, compression, and torsion—started almost at the same instant.

Into Ohio Special Quality Tubing goes all the skill, experience and painstaking care of an organization which for more than forty years has maintained on the making of the finest steel tubing. It is the tubing that has helped make possible many of the most advanced designs in planes that lead the world in speed, endurance and reliable performance.

Does tubing requirements today for speed-advance aircraft designers?

For today's size standards on "Ohio Special Quality Chromium-Plated Stainless Tubing"



The Engine Mount exemplifies the Ohio Special Quality Seamless Tubing used in the most advanced Curtiss Hawk 75-A Pursuit Plane. Tubing and fittings weigh 41 lbs.



CHROME
MOLYBDENUM
FOR
STRENGTH
LIGHTNESS
AND
SAFETY

The **OHIO**
SEAMLESS TUBE CO.
COLUMBUS, OHIO
Shelby, OHIO

AVIATION
June, 1935

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AC for Aircraft

Continued from page 37

is small. Because most wire used in airplanes has insulation sufficient to allow its use for 120 but not for 220 volts and because of the greater danger to quaking personnel if the voltage were 220, a voltage in the vicinity of 120 seemed advisable.

When the above data were presented to the A. C. Subcommittee, two members suggested that 22 to 46 volts would be more suitably optional. The findings of these men were based on a study of smaller airplanes, particularly where storage batteries were used, and their statement probably holds for this case. As previously noted, for commercial flying the airlines do not consider electrical systems necessary and were considering the use of electrical systems for use on only large airplanes. In view of these facts the only argument for a system having less than 120 volts was the thought that since the elements for 115-volt lamps are smaller in diameter than those of 12-volt lamps, they would be unable to withstand turbine vibration. A search of actual 115-volt bulbs were placed on the engine by alternately being burned for 200 hours, then started in an airplane for a flying time of 200 hours. It was found that bulbs averaging in size from 6 to 20 mm. in size on the average of 1000 hours of life. There was apparently no correlation between filament size and life. The 6-volt bulbs had an average life of nearly 2000 hours. Apparently, lamp life was more a function of the method of supporting the filament than filament diameter.

Discussion of Frequency

The frequency of an A. C. system has a direct bearing on the pounds per pound weight of a transformer. The higher the frequency the lighter will be the weight of the transformer used for supplying radio apparatus, navigation lights, voltage, and heater and lights. This phenomenon is shown as Figure 2. The weight of a 400-cycle transformer is considered as 100% in Figure 2, and it seems to be about 50% for 400 cycles, 25% for 600 cycles, and 14% for 1200 cycles. Lower weight transformers would probably be obtained if the frequency were further increased. Neglecting any reference to motion, for the time being, arguments were offered for 270, 360, 400, and later 400 cycles. The arguments against 600 cycles, the

proposed frequency of the group which would give the lightest transformer weights, were as follows: 1—Transformer efficiency would be poor at this frequency; 2—Because the lowest rate is very sensitive to an 800-cycle tone, it would be impossible to operate a radio receiver's phase and heater without undue harm to the receiver corpus. The first argument was answered by the curves given in Figure 4. The curve appeared in a paper by H. F. M. Alexander and describes some characteristics, the efficiency of which decreased, not increased, with increasing frequency up to 20,000 cycles. The second argument was answered by maintaining radio receivers with 400-cycle phase and heater power supply. A measurement of the loss from these supplies showed values of 40 db below 8 kilowatts, a value much less than that obtained for a dynamotor phase supply and a heater heater supply.

Directly, frequency cannot be said to affect motor weight. For a given motor type, weight per horse-power output is a function of its shaft speed. The synchronous (maximum) speed of a 400-cycle motor is 2400 R.P.M., of an 600-cycle motor 4000 R.P.M. If both types of motors were to be operated at their maximum speeds, the 600-cycle motor would be lighter than the 400-cycle motor, but if an 800-cycle motor were designed to operate at 2400 R.P.M., other things being equal, both the 400- and 600-cycle motors would have approximately the same weight. The other indirect relation existing between motor weight and frequency is that for a given induction, a condenser used for correcting the power factor of a three-phase motor, or for correcting the second-phase voltage at a single-phase motor must have twice the capacity and hence nearly twice the weight at 400 cycles than it has at 600 cycles.

A relationship exists between frequency and phase which will be discussed later.

Phase Discussion

The choice of phase affects the motor and distribution system. When more phases are used by a distribution system, more switches, fuses, and wires are necessary. While the total

weight of copper is the same for the same distance of transmission, some amount of power is to be transmitted, a given per cent loss, and the same voltage is neutral for a two-wire single phase, a four-wire two-phase, or a three-wire three-phase system, yet the total weight of conductors for a multi-phase system will be greater because of greater amount of insulation and larger diameter conductors required. For example, the weight of insulation on a three-phase system will be about twice that on a two-phase system, and the weight of conductors for a single-phase system is it is convenient to use the simplest structure for a ground distribution system, but that economy for keeping a three-phase system properly balanced offsets the use of three conductors.

These effects motor weights as shown on Figure 3. The figure shows that if the weight of a single-phase motor is considered as 100%, the weight of a similar two-phase motor will be 55%, of a three-phase motor 32%, and a D. C. motor 26%. If a motor is designed for three-phase, it can follow conventional spiral-type motor design, but if designed for single-phase, it had best be a "two-pole" single-phase or condenser motor. A condenser motor is, in fact, a two-phase motor which allows its other phase from a condenser in series with the first phase. These designs are the most practical since small high-speed motors with wound rotors that start without high mechanical stresses are difficult to build.

In order to secure high torque in a condenser motor, it is necessary to use low resistance conductors, and, as previously stated, the higher the frequency the lower is the weight of a given condenser for any desired capacity, hence, if a condenser motor is to be used, the frequency had best be more than 600 cycles. If it is chosen to have a frequency of 500 cycles or more, and if it is desired to use a conventional three-phase induction with good wire loss, an adequate number of slots per pole per phase must be used. If this machine had previously been designed at a lower frequency to have the maximum torque, the diameter, and hence the weight of the machine must be increased to have the higher frequency. If the supply is to be single-phase and high frequency, no motor alternative can be used to reduce. Such a machine has been built and has given splendid efficiency and a large number of watts per pound.

In view of the above discussion, I
(Continued on page 42)

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528 E. 12th St., P.O. Box 100, St. Louis, Mo.

AVIATION
June, 1935

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AVIATION RADIO

Dialing the Air Waves with Doc Fink



Airport Combination

A transmitter-receiver for airport use

THE HARRY RAND LABORATORIES, Cambridge, working in collaboration with the Boston-Maine Airways and Pennsylvania-Central Airlines, have recently brought out a combined transmitter-receiver for airports. The transmitter has 50 watts output power and operates, crystal control, anywhere in the region from 2,500 to 7,000 kc, either unmodulated, or modulated or frequency-modulated. The audio frequency circuit is simple, since it uses exactly a 40 crystal oscillator and a type 8C3B power amplifier. The modulator is a 500 cycle driven by two speech amplifiers. The tone generator is a 6007. The microphone is of the crystal type with a push-to-talk switch. The receiver is similar to the type 8C3 made by the National Company but

modified especially for this purpose. The frequency range includes the weather band, 200 to 400 kilocycles, as well as the intermediate and short-wave frequencies from 1,500 kc to 30 mc. The sensitivity is better than 1 microvolt, and the selectivity is 20 db down at 5 kc off resonance. The high frequency oscillator can be crystal-controlled at any two frequencies in the high frequency region. Two watts of audio output power are available.

The entire set weighs 10 lb and consumes 750 watts with full power output. The receiver and transmitter elements consume 280 watts. The equipment, as shown in the picture, is mounted in a rack cabinet approximately 67 in. high. The equipment is arranged so that it can be operated by one person in a standard telephone operator's booth. The tuning controls are located behind the front panel, and the only accessible control knobs are the single frequency control knob and the transmit-receive switch on the microphone unit.

125 Megacycle

Portable Radio Developed by Washington Institute of Technology

A NEW RANGE RADAR operating on 125 megacycles, and mounted in a trailer, is not only completely portable, but recently been developed by the engineers of the Washington Institute of Technology and delivered to the Civil Aeronautics Authority. The transmitter has an output power of 200 watts. A crystal whose frequency is 1/25th of the carrier frequency acts as the carrier wave source. The carrier frequency is modulated in an 800 cycle modulator oscillator, followed by two 800 cycle doubler stages. This brings the frequency to 400 megacycles, in which point it is amplified by two, in a pair of 10T 97 tubes push-pull. When the carrier frequency is reached a controlled driver stage employing 130 T9C feeds the output amplifier (two

200 T9C tubes). No modulation is introduced up to the output of the intermediate, but modulated modulation is employed in the antenna transmission line. Two frequencies of modulation are used, 50 cycles and 150 cycles, for the overlapping action of the range pattern.

The more transmitter and antenna supply is mounted self contained in a trailer which can be pulled by a 14-ton truck. The system contains the complete facilities including power generator, so as to be completely independent of all outside supplies. The antenna system, several forms of which are available depending upon the clearance



Range pattern of 100-Mc radar

of polarity and number of towers required, is mounted on a swivel frame at the top of the trailer. The modulating oscillators, antenna transmission lines, are driven by a self-contained synchronous motor.

The antenna elements are other placed and are mounted on ceramic insulators attached to the trailer support. Two types of antenna are of principal interest. One is a vertical parabolic flat corner antenna, consisting of four vertical rectangular plates arranged on the corners of a horizontal square. Horizontal polarization is used to obtain either a low-noise antenna or a two-corner arrangement. Other configurations can be arranged according to specified requirements, although the horizontal polarization gives the best results. The trailer which is fitted with a jack for parking purposes, can be handled readily by one man. An external power source can be used, if available, by means of connecting to the jack mounted under the trailer. It is possible to adjust the percentage modulation and the wave form of the modulating voltage

by increasing or decreasing the spacing between the plates in the modulator and oscillator.

The antenna system can be adjusted by varying the impedance matching elements. This adjustment is critical and has a marked effect upon the field pattern. Consequently, when any change is made the field pattern must be experimentally checked. A typical example of such a field strength measurement is given in the accompanying diagram. It is assumed that the transmission line external to the trailer, be kept completely straight in order to avoid reflections which are the pattern.

A generator capable of supplying power will operate approximately 15 hours on a single filling of fuel at full load. An important consideration is the effect of frequency regulation of the generator with respect to the modulation frequency. Since these frequencies are set by the synchronous

motor which drives the mechanical modulation, it is necessary that the supply frequency be held constant within narrow limits. The generator is equipped with a speed regulator which will hold the frequency within sufficiently narrow tolerances to operate with the receiving equipment using diode detectors. If the receiving equipment has tuned-circuit filters, then it is desirable to restrict the load variation on the generator to a minimum, in order to avoid frequency drift.

The transmitter can be adjusted for a given installation, usually upon acceptance of the equipment from the manufacturer, and it can then be depended upon to hold the calibration and adjustment even when moved over rough roads to another location. The use of ultra-high frequency carrier makes the entire installation independent of prevailing conditions and other factors.

AC for Aircraft?

(Continued from page 41)

before the following rule can be taken as essentially correct. If the acuity is to be three-phase, its frequency should be less than 600 cycles. If the acuity is to be single-phase, its frequency should be greater than 500 cycles.

The Case of the Discussion

After earlier discussions, the arguments centered around a 400 cycle three-phase system or an 800-cycle single-phase system, particularly with regard to the application of these systems to motors and what would be the relative weight, efficiency, and torque characteristics of motors for these systems. Since these discussions, a number of successful motors of both types have been constructed. The characteristics of two such motors are shown on Figure 5. The characteristics of these motors are not satisfactory, in the contrary, they offer a picture completely favorable with those given as typical for low frequency motors of the same size in the Standard Handbook for Electrical Engineers.

The weights for both type of motors were found to be nearly equal, but, anomalously, the single-phase motor was found to be a standard for 20 percent more than the three-phase motor.

Weight comparison by General Electric, Inc., for 1000-watt motors.

no such comparison. Looking at Figure 5, however, it will be seen that the three-phase motor has a power factor less than the single-phase motor, but the single-phase motor has a somewhat lower efficiency than the three-phase motor.

If the three-phase motor were corrected for power factor, its lower frequency, as previously stated, will demand higher weight conductors than those used with the conductor motor. The power plant for the single-phase motor must be made somewhat larger because of its lower efficiency, so we find that in the final analysis the weights seem equal for both cases.

This conclusion was hardly reached when the following argument occurred: "A large portion of the motors on an airplane, such as those used for landing gear, flap, ailerons, etc., are short duty motors and are not operated continuously, hence, we can use the three-phase motor and need not correct it for power factor. Furthermore, single-phase high-torque motors (of the short-duty motors) seem to be high-torque (also) require a centrifugal switch to change speed, and the switch is an undesirable feature."

We believe the above statement adequately expresses the only real advantage

of the three-phase over the single-phase system, while the statement previously made concerning transformer efficiency represents an advantage of the single-phase over the three-phase system.

From empirical data, the following equation was developed and it is believed suitable as a criterion for judging the relative advantages of the two systems for specific cases:

$$T_{3\phi} = T_{1\phi} \sqrt{1 + \frac{1}{2} \left(\frac{1}{\eta} - 1 \right)}$$

It is the above expression, the relationship which is greater than one, the weight advantage lies with the single-phase system; it has then one, with the three-phase system. The above equation does not take into account the relative advantages of a three-phase over a single-phase distribution system, nor the importance consideration of the distribution of this system which can be improved as follows:

$$T_{3\phi} = T_{1\phi} \sqrt{1 + \frac{1}{2} \left(\frac{1}{\eta} - 1 \right)}$$

As in the above expression, the relationship which is greater than one, the weight advantage lies with the single-phase system; it has then one, with the three-phase system. The above equation does not take into account the relative advantages of a three-phase over a single-phase distribution system, nor the importance consideration of the distribution of this system which can be improved as follows:

$$T_{3\phi} = T_{1\phi} \sqrt{1 + \frac{1}{2} \left(\frac{1}{\eta} - 1 \right)}$$

Advantage of 2 more weight, from

$$T_{3\phi} = T_{1\phi} \sqrt{1 + \frac{1}{2} \left(\frac{1}{\eta} - 1 \right)}$$

As in the above expression, the relationship which is greater than one, the weight advantage lies with the single-phase system; it has then one, with the three-phase system. The above equation does not take into account the relative advantages of a three-phase over a single-phase distribution system, nor the importance consideration of the distribution of this system which can be improved as follows:

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Berry Transmitter-Receiver

BUYER'S LOG BOOK

What's New in Accessories, Materials, Supplies, and Equipment

Tuned to market aircraft engine exhaust silencers and special exhaust manifolds, pipes, and collector air ducts, the Davis Aircraft Company has placed on the market an efficient and attractive exhaust silencer designed for various types of 40, 50 and 60 hp. engines. Materials are also constructed as replacement equipment for Hensons, Continental, Franklin and Lycoming engines in this power class.

The Davis exhaust silencer is of torpeda shape formed from lacemold, which retains a single flange indefinitely, adding materially to the appearance of a plane on which it is mounted. Acting as a combination of piezophone, the Davis silencer provides an excellent expansion chamber for the exhaust gases and also mixes these with cold air from the atmosphere by means of a vortex about which forms an eddy through the center of the torpeda exhaust silencer. At the rear of the silencer the gases pass out through a steel mesh cone which further eliminates both heat and noise. Flows at angle to flanges are visible from the exhaust end of the silencer.—*AVIATION, June, 1939*

A new type of dash mounting fastener having many aircraft applications has been developed by the Camber Fastener Company, 121 Elm 42nd Street, New York. Among the advantages emphasized by the manufacturer are simplicity of assembly, low weight, compact dimensions, improved aerodynamic properties, strength, and ease of maintenance. Weight of complete fastener is less than .65 lb.—*AVIATION, June, 1939.*

Capitometer when a new line of DC electric pressure gauges which are the result of several years of intensive experimental and research work. There are three different transmitters, one for fuel pressure, one for oil pressure, and one for manifold pressure. The electrical resistance transducer in all three units are similar. This interchangeable feature simplifies the maintenance and spare parts problem. The design, simplicity and accuracy of manufacture of the transmitters is such as to make it unnecessary to provide individually calibrated indicators. For example, any oil pressure transmitter can be used with any oil pressure indicator, any fuel pressure transmitter with any fuel pressure indicator, and any manifold pressure transmitter with any manifold pressure indicator. Indicators may be furnished with or without indirect lighting. The transmitter can be mounted in any position near the engine.—*AVIATION, June, 1939*

New fabricating facilities for beryllium alloys have been announced by The Beryllium Corp. of Pennsylvania. Heteroblastic beryllium alloys have been supplied commercially only as cast or upset bars. Now beryllium copper, beryllium nickel-copper, beryllium-chromium-copper, beryllium-nickel, and other alloys are available in the primary fabricated shapes of heat-treated rod, strip and wire.—*AVIATION, June, 1939.*

To save cost medium and small steel dies, punches, stripper plates and other miscellaneous parts, an open end lead wire diegrinding Model C6-14 has been developed by Carb. Sanders, of Graham, Wis. The many advantages of this new tool make it particularly desirable for aircraft die work. The set up for lateral grinding requires only 15 seconds. The die is removed from one opening into another, ready for grinding, with equal ease. The machine employs two blades of 4 in., 4 in., and 4 in. width. All use blades are 150 lb. long and are wound helically on a drive with a threaded groove.—*AVIATION, June, 1939.*

Just as a tough B&B log book is required to record as soon time through a busy harbor—as the Oliver "70" motor tugboat past TWA "Sky Club" planes at Stennis City airport. TWA has purchased a fleet of seven high compression "70" tugboats from Oliver Farm Equipment for service at four of the largest airports.—*AVIATION, June, 1939*



Davis Exhaust Silencer



Camber Fastener



Diegrinder between and Transducer



Oliver "70" Motor



K-Line Indicator



Carb Sanders New Model C6-14

A new bearing indicator of simple design for use in radio navigation has been developed by A. D. Moncrieff of the Civil Aeronautics Authority, and is being marketed by the D-Line Course & Bearing Indicator Co., at Chicago, Ill. The D-Line Indicator gives quick reference for any desired line, avoids all such problems as wind drift, ground direction and drift of drift, bearing from radio station to field, and other procedure problems. An ingenious feature of the instrument is a miniature airplane that moves with the chart to show the correct position of the plane and heading at any stage of navigation procedure. A large number of radio range charts are furnished with each instrument to standard equipment, and are contained up to date at all times.—*AVIATION, June, 1939*

A standardized strength stress loading system based on extensive experience with custom loading of various types of special loading equipment for aircraft, in conjunction with an specialized manufacturer of exhaust manifold rings, is announced by the Sider Aircraft Co., San Diego, Calif. It is to be marketed under the trade name "AIRMAX".

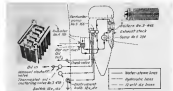
The new loader of extremely simple design, which has demonstrated its ability to be closed under severe conditions. Such loader units already having thousands of hours of service, are still in excellent condition. This loader is of "flush" type and is installed in the exhaust manifold at exhaust tidings. With a generating capacity of 40,000 H.P. in one hour, the loader has a weight of 340 lb., a length of 75 in., and diameter of 3 in. As many as four loader units may be connected in series for large bearing requirements. Working in conjunction with the Airmax boiler it is a special rubber of 30,000 H.P. U.S. The entire Airmax system has been made fully automatic and referred to a single electric switch for turning the system on or off. Yet the weight of a complete Airmax system, at 40,000 H.P. capacity, including water is only 21 lb.

However, in order to reduce maintenance, the Airmax loading system is completely self-actuated and provided with a special pump into which the water drains when the engine is shut off. The feed-water pump to supply the boiler is driven from any hydraulic or supply with as fast as the gas plus or returning loading gas system, ranging from 125 to 1360 lbs. per sq. in.—*AVIATION, June, 1939.*

FRANK PULS SERVICE AND MAINTENANCE MANUAL (in Spanish)—A 30-page booklet devoted to servicing methods and testing equipment for service and maintenance. B. G. Corporation, 116 W. 52nd St., New York

LO-SHOCK DASH PANELS FOR BLUE ATTACHMENTS—A 15-page booklet containing specifications and applications for Lo-Shock fasteners in aircraft work. Dell Manufacturing Co., Cleveland, O.

CANALAS No. 11—Convert hand driven and power driven for Phillips screws and power driven hand operated head drivers. Of particular value to see plan this a driving either standard or Phillips head screws with power tools. The Apex Machine and Tool Co., Dayton, Ohio



Transformer, Motor and Assembly of Alcoa heater

SAFETY

TO THE 'NTH DEGREE



THE WACO AIRCRAFT CO., TROY, OHIO

AVIATION
June 1939
91

THE AVIATION

NEWS

REVIEW COMMENT FORECAST

DANIEL BAYNE
C. P. McLaughlin, Public Const.
Wm. T. Sullivan
E. H. Lickie New York

JUNE 1939

P.A.A. Takes Lead on Atlantic

(Play in page 12)



REUTERS

WIRE PHOTO

RESERVED FROM MOSCOW RECOVER AT MISBOU: General Krasnodar and his company wreckage of the big red plane which he and Major Garmichev flew on last night from Moscow to the Gulf of St. Lawrence only to be forced to make an emergency landing on nearby Misbo Island. (Story on Page 12)



THE FLEETS GO! The Navy's aircraft carrier Ranger steams into New York Harbor with the rest of the Atlantic Fleet to help give a send-off to the World's Fair.

ANOTHER GERMAN has made another world's record for airplane speed over a thousand-meter course, according to reports from Berlin. This time the hero is 24-year-old pilot Wernke. The plane is the Messerschmitt 109 B and the speed 753.71 km. per hr., about 468 mph.

(Play on page 12)



AVIATION PEOPLE



PAINTER: Leslie Shaw, perhaps the world's foremost artist specializing in aerial subjects, is now in the U. S. in an official visit for the Air-Wildlife and Airm-Globe of France.



TWO OF A KIND: John Wagner, chief pilot for Vultee (left) plane with Edward T. Klein, immediately known test pilot and accounting engineer, who has joined the Boeing Aircraft Company as Chief of the Research Division. First job, the mounting of tests on the Boeing Model 200 standard to help in the record of those ships late last month.



PAN AMERICAN: John M. Franklin, president of the U. S. Lines and the International Maritime Marine Co., has been elected a director of Pan American Airways.



NEEDS: Arthur L. Low, now will be the chief director at the National Academy of Sciences, which opens at Chicago, June 10. Adams, senior, previous the U.S. and last month.



ON LOAN: Ray O. Myers, borrowed from United Airlines has been loaned to Santa Monica, leaving the 80-4 and the DGA. The DGA was due to be turned over to the airline late last month.



TRAVELING SALESMAN: Carl E. Spicer, vice-president, in charge of sales for Lockheed has left for a demonstration flight with the new Lockheed 14, to Latin America.



ALSO PAN AMERICAN: Thomas A. Morgan, president and director of the Navy Corporation, has been elected a director of Pan American Airways.



DIRECTOR: Frederick W. Orr, assistant general manager of Douglas Aircraft and a veteran engineer specializing in aircraft construction, has been named a director.



FAIRMARE: G. A. Penick will take on at the aviation building at the New York World's Fair. He is general manager of U. S. Aviation Exhibits, Inc.



TEN YEAR MAN: W. A. Peltzer, having just five years as U.S. president, ten years as executive manager, was named an honorary director of the staff.



BROOK STREET: George H. Dwyer, head of Engineer's Aircraft Company, Ltd., is now in the U. S. to demonstrate his special aircraft as installed in a Ryan 8C.



LYCOMING "50"

The Star of the Skypways

"Quiet and smooth" . . . "Efficient power" . . . "Given pilot instant confidence" . . . "Runs quickly without fuss" . . . "Aircraft from ordinary every-day, peeping engines" . . . "A conscious flow of smooth power as one chafes at the start" . . . "Get competence before any competitors" . . . "Perform like a warrior" . . . "Better performance than any other '50'" . . . "David's best self" . . . "These are just a few of scores of frank expressions from Lycoming '50' owners. Try behind 'The Star of the Skypways' and feel the difference!"



FREE FOLDER

This illustrated folder gives you complete details and specifications of the popular Lycoming 50 HP aircraft engine. Ask your Taylorcraft, Cessna or American Builder for your free copy or mail us. Address: Department AF-1.

Let's Take a Look at the Records!

You hear a lot of talk about 50-horsepower engines these days. But as far as the Lycoming "50" goes, you don't have to listen to *generalities*. Let's take a look at the *records* compiled at the end of the recent tour by the Lycoming Trio—a Taylorcraft, a Cessna and an Aeromac, each powered by a Lycoming "50"—with the pilot of each ship having his own itinerary and giving sight demonstrations to light aircraft prospects, in temperatures ranging from zero to 95° F.:

- ★ Total Number of States Covered 19
- ★ Total Number of Airports Visited 193
- ★ Total Number of Distributors Contacted 73
- ★ Total Number of Flight Demonstrations 1,105
- ★ Total Number of Takeoffs and Landings 1,110
- ★ Total Number of Flying Hours 711
- ★ Total Number of Miles Flown 61,987
- ★ Average Miles Covered per Hour 86.8
- ★ Total Gallons of Gasoline Used 1,971
- ★ Average Gallons of Gasoline per Hour 2.7
- ★ Average Miles per Gallon of Gasoline 22.4
- ★ Total Quarts of Oil Consumed 15
- ★ Average Quarts of Oil per Hour 0.021
- ★ Average Miles per Quart of Oil 4,133

LYCOMING DIVISION OF AVIATION MANUFACTURING CORPORATION, WILLIAMSPORT, PA., U. S. A.
Circle Address A4115104



Presenting Another KOLLSMAN DIRECTION INDICATOR



Actual Size

Four months ago the Kollsman Instrument Company announced an entirely new and original aircraft compass—the Kollsman Direction Indicator.

Now Kollsman is proud to announce a complete instrument designed primarily for use by the navigator rather than by the pilot. This new instrument is the Kollsman Direction Indicator with Horizontal Dial—Type 4410B. It is larger in size than Type 4410A.

and is graduated in single degrees, as here illustrated.

This indicator employs a dial corresponding to the compass rose printed on all aircraft charts, a pulsator, and variable reference lines. It is easy to read, having no parallel cover. It has a very long period and practically no creeping. It comes equipped with Poly-Phase compressor and Kollsman Rim Lights.

Further detailed information will be sent on request.

KOLLSMAN INSTRUMENT CO., Inc.
8008 FORTY-FIFTH AVENUE ELMHURST, NEW YORK

WESTERN BRANCH: GRAND CENTRAL AIR TERMINAL, OLYMPIA, CALIFORNIA

Factories Spread in Cal. Lockheed, Northrop, Douglas get new floor space

When Southern California aircraft factories opened at such high levels this month of May, many factory executives still are on edge of the day, for the Lockheed Aircraft Corp., whose 1934 factory additions had brought total floor space to over 500,000 sq. ft., a new machine and service building was being rushed to completion. Meanwhile approximately 100 by 100 ft. in size the new building adds some 20,000 sq. ft. to Lockheed's floor space. About the same amount of floor space is being provided by creation of a new three-story wing adjoining the other building at the head of the factory, to provide fully rented machine shop space. And while Lockheed continues its factory expansion program, so too will the Los Angeles Company, the huge Douglas factory between the state's expansion program by again expanding its physical dimensions in accordance with operations of the El Segundo Division. The new Douglas building has a 15 ft. clearance under ceiling and provides an additional 11,000 sq. ft. of floor space.

Meanwhile the new 180,000 sq. ft. factory for Northrop Aircraft, Inc., is being laid out in Long Beach Municipal Airport, and North Aircraft Corp. occupies the 82,000 sq. ft. factory in Los Angeles Municipal Airport. Van Noy, that was actually built in 1929 for the North Aircraft Company. Total employment of all aircraft factories in the Los Angeles County area, including all plants working on aircraft assembly, undoubtedly had reached 100,000 in May.

Meanwhile the new 180,000 sq. ft. factory for Northrop Aircraft, Inc., is being laid out in Long Beach Municipal Airport, and North Aircraft Corp. occupies the 82,000 sq. ft. factory in Los Angeles Municipal Airport. Van Noy, that was actually built in 1929 for the North Aircraft Company. Total employment of all aircraft factories in the Los Angeles County area, including all plants working on aircraft assembly, undoubtedly had reached 100,000 in May.

Douglas Hears 48-Million Aircraft Output; New Models

Somehow a year passed the following, the Douglas Aircraft Company completed and delivered 48,000,000 worth of Douglas aircraft were going down the period Dec. 3, 1934, April 16, 1935, Douglas Douglas, pres. said the annual production meeting on April 16th. "We had orders on April 16th worth \$22,204,000.00—A size of the DC-3 is

passenger transports in new order production, with an order for under planes of this model under production. Forty orders for the DC-3 were to be completed, Douglas thought, as the plane is now through the CAA tests and will shortly be in service test operations on the various airlines. As the meantime the Douglas company is working on the design of a DC-4 four engine plane of about the same capacity as the present DC-3. Douglas stated that his company expected to be manufacturing aircraft at the rate of 100,000,000 per year in the near future. Not in very long ago \$40,000,000 output per year was a great way from the figure for the 1935 aircraft industry.

Chamber Steps Out Sets up aggressive Export Committee to push sales

THE AERONAUTICAL CHAMBER OF COMMERCE OF America held the first meeting of its new general Export Committee in Washington May 2, and adopted a long term program intended to protect the industry's foreign trade and standards at whatever opportunity. John H. Jansin, president of the Chamber, opened the meeting and described some of the export problems confronting manufacturers of aviation equipment. John M. Rogers, vice president of the Douglas Aircraft Company, was elected permanent Chairman of the Committee.

During the all-day session it was agreed that steps should be taken looking to the improvement of export and country military conditions comprised of factory representatives; the services of those companies in the industry associated with those of the Chamber's Export Committee.

For the purpose of having a record on the economic importance of aviation exports the committee agreed that members be asked to report to the Chamber periodically the percentage of their gross business classified as export, and the amount of domestic equipment acquired by their exports during a given period.

The Committee also decided to set up an export financing



SOLAR DOUBLES—is expanding floor space by the construction of a new building at its Los Angeles factory site.



VAN Noy BUILDING—May 16 for a total new \$100,000 factory on west side of Long Beach. Factory now totals 180,000 sq. ft.



BESTER ORDERED—New aircraft production building from the Avia Company to house equipment and engineering departments.

making wing pumps. Headquarters of the Aircraft Research Corp. will remain in Glendale, where a line of hydraulic test wing equipment is made.

New Plane Company

Plans to produce a new type light aircraft in order phase of purchase contract, the North Pacific Aircraft Corporation has been organized in Seattle, Wash., by George Yelen, and Alton T. Greenwell. While no further

plans have been announced, the first plane, known as the North Pacific, is now being built from the plant is of conventional low wing design with retractable landing gear. Power is furnished by two 15-horsepower 16 hp engine units mounted in wing nacelles. Structures of the plane throughout in of spruce strips united in glue-bolted fashion over bulkheads and fuselage. Accommodations are provided for two people seated in tandem in an enclosed cabin.

Four Per Day at N. A.

Currently building airplanes at an unprecedented 4,000 units a month, North American Aviation Inc., Inglewood, Calif., reported sales aggregating \$4,000,000 for the quarter ending March 31. Last week, reported with \$100,000 for the like 1935 period, and a backlog in of March 31st of \$1,000,000. Net profits for the quarter were \$1,000,000 after all charges (costs to approximately 20 cents a share on stock outstanding). Production is currently at the rate of approximately four planes per working day, with heavier production concentrated in the home market, being delivered to England.

Six Hundred Workers

A new 200,000 sq ft factory will soon arrive to triple the facilities of the Aircraft Engineering Division of General Motors Corporation, at Indianapolis, Ind. Six hundred workers in 100 per cent automatic will produce Allison high-speed, V-type 12 cylinder, liquid-cooled motors, and a new 14-cylinder model the rights of which of the General Motors plant at the New York World's Fair assumed considerable interest.

Gains Security

American Aircraft Co. has been forced to take over the assets of the Security Aircraft Corp. of Long Beach. The new company will have an authorized capitalization of \$50,000, shares of preferred and \$100,000 shares of common stock, both of \$1 par value. Security Aircraft Corp. is to control 125,000 shares of common stock in return for its assets, which consist of a study of his Security Aircraft folding wing monoplane and the Security engine with which it is powered, both designed by W. H. Roman, president of Security Aircraft, and of the new line.

Expanded Expansion

Two May issues of AVIATION reported that the Wright-Bellanca Corp. at Bridgeport, Conn., would have an area expanded by 4,000 sq ft, plus an additional building of 400 x 350 ft. Since then a new contract has been awarded to increase the size of the new addition to 600 x 150 feet. With the completed addition the total construction will approach \$1,000,000. The added space, too, will accommodate the purchase of more equipment, and will probably push the completion date from

ward to August 1st, instead of July 1935, as originally planned.

Grumman Bookings

Three new new contracts for airplanes and parts totaling \$1,500,000 recently landed in its backlog, Grumman Aircraft Engineering Company, San Diego, Calif., announced in \$1,400,000 as of April 18. Deliveries for the next 12 months of 1935 were \$1,150,000, leaving \$1,410,000 for the 1936 period, a gain of 25 per cent.

To Build Plant in West

IMMEDIATE negotiation of a \$150,000 plant in the Los Angeles-Burbank area by the Cleveland Pneumatic Tool Company has been announced by James S. Butler, president of the Los Angeles Chapter of Engineers, following receipt of a visit from L. W. Lewis, president of the Cleveland concern. The new factory will be located in a 30-acre tract in the city of Burbank and will specialize in the manufacture of airplane parts, landing gears, and other pneumatic units.

Headwinds for Export

MAKING a strong bid for the export market, Howard Aircraft of Chicago has introduced a new version of the DGA, and has revised its foreign marketing plan. The DGA-1000 and DHC-1000 have been modified in certain detail to enhance their value as passenger and cargo aircraft in out-of-the-way places. They are adapted to wheel, bush or skid landing gear. They are first for maintenance or possible repairs. On the P-1000 series the engine is a P-1000 of 100 hp. On the DHC-1000 is a 100 hp. On the P-1000, a 100 hp. engine is standard. Both models are Howard Standard convertible props. At additional cost, full equipment for night or day flight is available. The former plan of marketing through an export agency has been dropped, and in the future Howard airplanes will be sold direct by an export division of the American Manufacturing Corp.

New Stinson Staff

AMERICAN MANUFACTURING CORP. announced two important additions to the staff of the Stinson Division at Wayne, Mich. Robert E. Galt, ex-chief engineer for Curtiss Wright at Buffalo, and will move to AVIATION. The added man, too, will assist in the production of more equipment, and will probably push the completion date from



Curtiss

THE FOREMOST BUILDER OF PURSUIT AIRCRAFT

● Curtiss has led the field in building pursuit aircraft for the U. S. Government since the first pursuit type, designated as the P-1, was developed for the U. S. Army Air Corps in 1913.

The latest pursuit order awarded Curtiss consisted of a large number of the newly designed Curtiss P-14 Type—top speed Standard Pursuit Airplane of the U. S. Army Air Corps—was recently accepted and construction is now under way on a service test order for Curtiss YP-17 high-altitude pursuit airplanes.

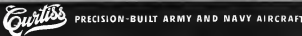
A large production order for planes of the Curtiss P-14 Type—the present Standard Pursuit Airplane of the U. S. Army Air Corps—was recently accepted and construction is now under way on a service test order for Curtiss YP-17 high-altitude pursuit airplanes.

For nearly a quarter of a century, Curtiss has been a prominent builder of aircraft for the United States Government and is now playing a major part in our nation's program of having an adequate Air Force for National Defense.

CURTISS AIRCRAFT DIVISION
CURTISS-WRIGHT CORPORATION
Buffalo New York



NOT TO BE CONFUSED—shortly after Pratt's disclosure of another big engine (see page 41), Wright Aircraft built the new of the (unpublished) Pratt & Whitney engine, a double row, six-cylinder, 18 cylinder model with a 2000 hp rating. It has 4 main cylinders, 1000 each, built in the Pratt & Whitney engine. Pratt's disclosure was in Generalissimo's long (page 41) type, last (page 41).



REPORT CARD

Of Air School

A new engineering building is going up at Purdue to accommodate an increasing number of students. Two rooms, each 612½ feet and each containing 40 individual drafting tables will be provided. Nonparel fluorescent lighting is specified for these two rooms. To increase the usefulness of the rooms, lecture platforms and blackboards are to be provided. A third room of equal size is to be used as a lecture and motion picture room, while a fourth will be an engineering laboratory. In the laboratory is to be a concrete base built for use in conducting load tests. It is planned to install the most modern equipment throughout the laboratory.

Leading school scholarship winners of 1934-35 in aerodymics made recently by the National Committee of Award of the United Air Lines Scholarship have been announced. Four awards were given, consisting of the Solo Pilot Flight Course, plus three others of one of the following courses for which the candidates were eligible: Air Transport Engineering, Airline Operations, and Engineering, Airline Techniques, Airline Meteorology, Airline Mechanics, or Airline Maintenance, and Airline Maintenance Course, Wisconsin.



HUNSAKER SCHOOL—Recently dedicated was the newest and largest quarters for the Hunsaker School of Aeronautics, at the West Branchford drive, Philadelphia, according to the recent announcement by President Ford W. Hunsaker. The new home is a three-story building having 45,000 sq. ft. of floor space.

where: Louis Compton, Pittsburgh, Pa.; Richard B. Duff, New York, N. Y.; J. Paul Kessel, New York, N. Y.; and Stewart W. Day, Atlanta, Georgia. These Young Scholarships are awarded annually for the best student on an unrecruited subject by a student regularly enrolled and in good standing in a college, university, or junior college in the United States or Canada. The 1934 National Committee of Award was composed of: Dr. Eddison H. Woods, University of California, William H. Reed, Stuart Research Laboratories, Philadelphia; Bradley M. Zinn, University of Minnesota, and A. F. Brunschwig, Aero Dept of Flying, United Air Lines.

The American School of Aircraft Engineers, at Glendale, Calif., enters its flying semester with a maintenance curriculum. The school offers a complete maintenance course in aerodynamics, repair and testing of all instruments, check assembling, panel design and lay-out, plumbing and wiring. Aid in instruction is under the personal supervision of Gene C. DeLeon, Government. Approved instrument instructor, formerly with the Illinois and Consolidated Aircraft Corporations.

College examinations are now planned. The tests at the school are to be given by the students learning to fly under the C.A.A. experimental training program. Those with successful results will be given a heavy load to carry. A check-out will be expected in June 1935.



C.A.A. TRAINING POST: Hanger and flight line at the Airplane Institute of Aeronautics, Yonkers.

C.A.A. Flight Training Log As Of May 15

Name	Total Hours Logged to Date	Number of Students	Cost Record
University of Arizona	100	10	10
University of California	100	10	10
University of Illinois	100	10	10
University of Michigan	100	10	10
University of Minnesota	100	10	10
University of Wisconsin	100	10	10
University of Texas	100	10	10
University of Oregon	100	10	10
University of Washington	100	10	10
University of Colorado	100	10	10
University of Nevada	100	10	10
University of Idaho	100	10	10
University of Montana	100	10	10
University of Wyoming	100	10	10
University of Utah	100	10	10
University of Arizona	100	10	10
University of California	100	10	10
University of Illinois	100	10	10
University of Michigan	100	10	10
University of Minnesota	100	10	10
University of Wisconsin	100	10	10
University of Texas	100	10	10
University of Oregon	100	10	10
University of Washington	100	10	10
University of Colorado	100	10	10
University of Nevada	100	10	10
University of Idaho	100	10	10
University of Montana	100	10	10
University of Wyoming	100	10	10
University of Utah	100	10	10
Total	100	100	10

during this period, it is sure to be followed by a rapid escalation after the flying days have passed.

Two who were working in two weeks by taking the course of the Hunsaker (Mich.) School of Aeronautics. Classes begin July 15, July 16, August 15. A four-week course is also offered beginning July 15. Participants learn and help out for the two weeks courses in \$140. For the four weeks course, \$225.

Two instructors are needed by the Hunsaker High School of Aeronautics. James New York City. The vacancies are for an instructor in aerodynamics and a teaching assistant. Both for at least one semester that is from September 1934 to February 1935. Temporary salary is \$125 per month but those who qualify themselves to take the regular course examinations are eligible for a salary schedule beginning at \$140 per month to a maximum of \$200. Requirements are high school education, 4 years of aerodynamic experience in the industry, and aviation instructor's training course.

Aircraft power plant operators is the subject of a course of instruction by the Webb Aircraft Repair School of Technology at Glendale, Calif. Executive personnel include: D. F. Webb, president; L. C. Webb, secretary-treasurer; and Charles A. Brown, controller. Also in-

structors are James Goodrich and George H. Eberhart, president Webb was associated with the Douglas Company when he prepared to found the school.

Over a thousand books have been distributed by the Hunsaker School of Aeronautics to schools and public libraries recently. These free books are being supplied by the Hunsaker School of Aeronautics to schools and public libraries recently.

With all the talk about building up a defense air reserve for military purposes in fact, even so the C.A.A. pilot training program will be high gear, in preparing to drive a substantial program under way in Southern California as indicated with considerable emphasis. Since in the California Aero Squadron, and sponsored by the Hollywood Junior Chamber of Commerce, the group meets regularly at Los Angeles Municipal Airport, training is the first significant development in the industry, and aviation instructor's training course.

BERRYLOID AIRCRAFT FINISHES PROTECT U.S. NAVY'S NEWEST DIVE-BOMBER... CURTISS SBC-4... INSIDE AND OUT



ANOTHER 100% BERRYLOID JOB

Outstanding in ability and performance of their intended function, the U. S. Navy's new Curtiss SBC-4 squadrons will soon be among the Navy's most formidable groups of Dive-Bombers.

Fittingly enough this new product of "The Planners of Aviation" is finished throughout in another pioneer aviation product—Berryloid Aircraft Finishes, for beauty and protection and genuine economy.

Write
BERRY BROTHERS
ST. LOUIS, MISSOURI
BERRYLOID DIVISION



Wings on the Wind

(Continued from page 25)

lent the English States-Cable and the German Huns.

In view of the impossibility of combining all desirable characteristics in one instrument, most airplanes participating in important contests carry two instruments: one for precision work and one for prompt indication.

As for tapered meters, the German Airforce produced a remarkable instrument in 1938. After many years of campaigning the use of Pitot and Venturi tubes, Airforce has come to a simple Venturi tube type. Accurate speed indication is given from 20 mph to 125 mph. Greatest precision is obtained in the range between 25 mph and 50 mph. The diameter of the instrument is 2 in. The Venturi tube is electrically heated to prevent freezing.

Venturi tube type indicators have generally been recognized as unsatisfactory for soaring flight because they frequently go out (due to ice and water accumulation) at the time when they are most needed, that is, when flying blind in a storm cloud. Very light weight electric type indicators have therefore been developed which are driven by pocket flashlight batteries and are switched on only when actually needed.

Excess in Finland where long winters produce ideal solo-out conditions from the ice-covered surfaces of lakes, automobile towing has been almost all of its popularity. Windy towing is very much in favor everywhere. The two-placed, sailplane design arrangement found in the rear wheels of canopies has generally disappeared and in place has come the standard wing which is generally mounted inside a car inside the driver's seat. A double chair behind the gear has connects the engine to the rear wheels for normal driving or with the chain transmission on the wing for launching. Wind drums are usually of steel and must be perfectly balanced to avoid vibration. For school purposes, towing ropes are generally 3/4 in. diameter steel cables (34 wires with long ends) approximately 3,000 ft. long. A safety system is usually installed in the rope which will break under a tension of 300 lb.

The drum diameter and the transmission are calculated for an average speed of 40 mph in second gear. First gear is employed only for normal per-

formance like pulling a ship up from a valley to a hilltop, or doing other heavy work. Third, or high gear, is used only when it is absolutely necessary to launch a sailplane with a tail-wind, or to wind the rope up rapidly to get it out of the way of a plane which may be landing on the field.

This type of wind was developed in Milan, Italy, and was successfully used at the Olympic Soaring meet in Berlin in 1936. Advantages are obvious. The cable which is independent of other launching means and on other use it needed on the field.

AIRCRAFT towlines are now very popular in Europe and many pilots of several schools have tried to solve the problem of what to do with the towing rope after the airplane has released it. Normally, the airplane pilot glides toward the field, passes over it at a height of about 100 ft., and pulls his cable without to drop the rope up the field. He then has to make another turn around the field in order to land. This procedure can be avoided by pulling the cable inside the airplane, that is providing a small winch inside the ship to retract the towing cable after the glider has cut off. In order not to introduce undue complication, the towing line does not come directly on the wing but is normally taken up through the attachment fittings at the tail of the airplane. The cable release is operated in the usual way but instead of the main cable dropping freely, it is pulled up by a short length of cable which is in the release rig and the work done. After pulling the release, the pilot immediately starts winding the cable on board. As he does not require his engine running, he may be surprised for his landing. In the meantime, this saving the necessity of making a second turn around the field. Thus between take-off it released and an appreciable amount of gasoline may be saved. At the same time, the dangers and inconvenience of dropping the cable into the field, later to be located and re-traced are eliminated.

Due to the interest in modern altitude and distance flights, most secondary schools have gone in for remote blind flying training. The German research center at Darmstadt has probably gone further in this direction than most schools. A Kratoch two-seat sailplane is used with complete instrument provided for use of the nose. Control are usually held during the winter when the air is comparatively calm and when normal soaring activities are at a minimum. An airplane

pulls the ship to an altitude of 10,000 ft and the student pilot then has an opportunity at blind gliding. Where conditions are lacking, the same thing may be done with temporary balloons around the pilot which may be lifted or well.

A GREAT DEAL of consideration is now being given to safety. It makes little sense that a pilot should be injured in a landing which is simply rough. Safety belts should be worn because in rubber seat pressure on the body to a noticeable limit and no accident should be permitted in front of the legs and knees instant which the pilot should be thrown in case of a sudden stop. It is particularly important that the modern indicators be easily detachable. The pilot should be able to see it and possibly throw it away in any possible accident of the ship. Hinges should not be on one side and the release on the other. In spite of all precautions, it has happened more than once that an indication in a spin or in a tight diving spiral have kept the instrument from coming to permit the pilot leaving the plane by parachute. The more practical there is to make hinges detachable so that every hinge can be used as a release.

THE YEAR 1939 will see a great advance in two-seater in European soaring circles. They are beginning to invade the field of remote aerobics removed the single-seater. The only sailplane which crossed the Alps twice in a two-seater. The distance record for two-seaters shown here passed to 252 miles. The absolute record for duration was set at 31 hours by a two-seater as compared with a 40-hour duration record for single-seater. Good recently aerodynamic and structural problems hampered the development of two-seaters but with the development of the modern wing, with aerodynamic improvements on control, and aerodynamic improvements in rubber, the two-seaters have been made almost the equivalent of the best single-seaters in performance.

Success stories in Europe may be summed up to include the use of thinner wing sections, lower wing loading, greater comfort and safety for pilot, and better instrumentation. Sailplanes entered the high altitude work must be very strong and have greater stability than normal sailplanes. They must be fitted with electrically heated and electrically operated instruments, with oxygen equipment, and carry protection against control failure through freezing.

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time and space...a

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FLYING BOAT

charts new paths for
AMERICAN EXPORT
AIRLINES



CONSOLIDATED *Aircraft* **CORP.**
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America's Answer

(Continued from page 28)

the monthly demands within the capacity of 1,000 men. This can then readily be expanded to cover the average number of men who will be stationed in the two years of the expansion program. By this method, we see that under full capacity conditions, more than the required 30,000 airplanes can be produced to meet the airplane industry will only be at about twenty-one per cent capacity. For the engine-propeller portion of the industry, the percentage of capacity will be twenty-one per cent. As men-

tioned before, these figures are desirable to allow for unforeseen contingencies. They are, nevertheless, gratifying as an indication that extensive base space expansion is not required to do the job.

It is revealed that every unit of the industry be kept in approximately the same percentage of its capacity so that overloading of isolated units will not result, leaving them to over-expansion. Sub-contracting within the industry would be a most desirable procedure in such contingency.

Here, however, production capacity, it is now desirable to study appropriations from the standpoint of their adequacy for the present coupled with estimates of probable requirements for the next fiscal year. In Table 11, for the Army, Navy, export and trans-

port markets, the average weight of planes (previously determined) is multiplied by the quantities required and the total given to give a total cost. This process is repeated for engines and instruments, providing determinations of the grand total required in each of these fields. The results, it should be remembered, cover the two year period involved in the expansion program.

To the layman, at least, there is always considerable confusion in trying to determine from appraisals submitted the exact funds available for the procurement of aircraft. These are generally hidden in many appraisals in the form of "contingencies" which frequently are not needed as it is made it well-known impossible to determine just what proportion is for new construction, at what may be accounted for existing, maintenance, or other use. From such sources as are available, it appears that the situation as now being considered by Congress is about as shown in Table 12, where the funds are divided between "available" appropriations and "general" funds appropriated for the use of the air force. An indication of probable required appropriations for the first year when the present one now being debated in Congress is worked out on the basis of funds needed (as determined in Table 11). From this determination, it would appear that the expansion of the Army air force, the full \$300,000,000 originally mentioned for this purpose by the President should be made available for the exclusive procurement of aircraft. If \$170,000,000 only is allowed for the purpose, as recently indicated may be the case, the expenditure in the second year of the program will naturally have to be considerably higher than would otherwise be necessary, as this should be necessary for the "expansion" appropriation bill for that year.

When making a comprehensive analysis of this type, it is desirable frequently to check up by other methods than the broadly used, in order to be sure that conclusions are approximately correct when viewed from other angles. A check on funds required is made in Table 13 by studying labor output. First there appear figures showing the airplane and engine rates-per-dollar sub-divisions. Then, by determining the average yearly wage of the workmen and dividing by the percentage that direct labor bears to the total value of the airplane or engine, a figure is determined which shows the dollar value of airplanes and

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Table 8 Production Capacity—Airplane

	Men	Planes Per Month	Planes Per Year	Planes Per Month	Planes Per Year
Actual Accomplishment	5,000	4,750	57,000	4,750	57,000
Army — Average	1,000	4,750	57,000	4,750	57,000
Navy — Average	1,000	4,750	57,000	4,750	57,000
Export — Average	1,000	4,750	57,000	4,750	57,000
Transportation — Average	1,000	4,750	57,000	4,750	57,000
Aviation — Alaska (Dept. of Interior)	1,000	4,750	57,000	4,750	57,000

U. S. Airplane Industry—Military, Export and Transport

	Men	Planes Per Month	Planes Per Year	Planes Per Month	Planes Per Year
Present Program	5,000	4,750	57,000	4,750	57,000
Army — Average	1,000	4,750	57,000	4,750	57,000
Navy — Average	1,000	4,750	57,000	4,750	57,000
Export — Average	1,000	4,750	57,000	4,750	57,000
Transportation — Average	1,000	4,750	57,000	4,750	57,000
Aviation — Alaska (Dept. of Interior)	1,000	4,750	57,000	4,750	57,000

Table 10 Production Capacity—Engines and Propellers

	Men	Engines Per Month	Engines Per Year	Engines Per Month	Engines Per Year
Actual Accomplishment	5,000	4,750	57,000	4,750	57,000
Army — Average	1,000	4,750	57,000	4,750	57,000
Navy — Average	1,000	4,750	57,000	4,750	57,000
Export — Average	1,000	4,750	57,000	4,750	57,000
Transportation — Average	1,000	4,750	57,000	4,750	57,000
Aviation — Alaska (Dept. of Interior)	1,000	4,750	57,000	4,750	57,000

U. S. Aircraft Engine and Propeller Industry—Military, Export and Transport

	Men	Engines Per Month	Engines Per Year	Engines Per Month	Engines Per Year
Present Program	5,000	4,750	57,000	4,750	57,000
Army — Average	1,000	4,750	57,000	4,750	57,000
Navy — Average	1,000	4,750	57,000	4,750	57,000
Export — Average	1,000	4,750	57,000	4,750	57,000
Transportation — Average	1,000	4,750	57,000	4,750	57,000
Aviation — Alaska (Dept. of Interior)	1,000	4,750	57,000	4,750	57,000

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MODERN AS A FOUR ENGINE POWER, SAFE IN A STALL... SAFE ON THE GROUND... THE HARLOW ALL-METAL TRAINER... WITH RETRACTABLE LANDING GEAR AND FLAPS DISCONNECTED IN 30 SECONDS... FLAPS WITH ONLY A RUDDER, STICK AND THROTTLE TO OPERATE. AFTER SOGO... AND BY EASY STAGES... HE CAN ACQUIRE THE USE OF REAR EFFECTIVE FLAPS... THE EFFICIENCY OF A RETRACTABLE LANDING GEAR... AND THE FEEL OF A GIGAWATT PERFORMANCE AIRPLANE.

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(Continued from page 20)

profits from another ship, represents but shows that an airplane company will expend sixty per cent of its costs on a contract at the time the first plane is delivered when the total quantity under order is fifty planes. The percentage changes to forty when the quantity is one hundred, to thirty-three when the quantity is one hundred and fifty, and to twenty-five when the quantity is three hundred. These percentages are indicative of the time lag between receiving a contract and building substantial and fairly constant monthly deliveries.

The foregoing applies to what may be termed "current" models. However, planes are ordered from blueprints or from one prototype, as in our experience to equip our squadrons with the most up-to-date type; may be the case to some degree in any expansion period, then it may be anticipated that on the average, order months will elapse before regular production deliveries are received. The necessary time is to use for preparation of production drawings. Naturally, this figure varies, depending upon the size of the airplane involved but will always be between ten and fourteen months. Let us assume for this case that twelve months after a contract is placed, deliveries will reach ninety-five airplanes per month. Now, if the order is for three hundred planes of

an entirely new type, it will be twenty-four months before the last ship is delivered, as, twelve months allowed for obtaining complete delivery plus twelve months to complete the contract at delivery rate of twenty-five planes per month.

Although in many cases, orders will exceed three hundred, reaching five or six hundred for certain classes, it should be realized that in general, the average for all classes, that is Army, Navy, export and transport, will be about two hundred and twenty airplanes. This can be readily checked as follows: There are about fifteen classes of Army planes, ten classes of Navy planes, and five of transports, with deliveries in some classes, making approximately thirty-two distinct types, being produced at one time. When the total of 9,300 airplanes required in two years is divided by thirty-two, the answer is approximately two hundred and twenty for the average quantity in each order. All at this means that most of our order books, getting under way and getting contracts actually placed at an early date is the most important single factor involved in obtaining the full strength of air forces which we have set out to realize by the required date, July 1, 1941.

It is interesting to note, relatively, the yearly production rates of the past, at the present, and of the future and

first which currently maximum for certain foreign countries. A final indication, Table 15, shows this information. Amounts by dollar value for various periods of time are indicated along with the industry, average unit price of airplane, and number produced per year. Roughly speaking, in terms of dollar value, it is seen that if 1938 production is taken at 100, then the average production of the four years previous to that was 55, whereas for the next two years it will be 262, leading down to 189 thereafter. While no exact picture of the total two years as average of 4,832 airplanes per year, it is estimated that England is now producing at the rate of 7,000 planes per year and Germany 12,000.

What then does it all mean? First, the industry can definitely do the job with only a very small additional "bolting" expansion of about ten per cent in four years.

This being the case, it is extremely important that our government even expansion. Now is the time to consider supplying the links. World-wide demand for military strength is at a high level and must inevitably fall off at some time in the future. Let us not forget the parallel of 1928. A month from a month ago, as President Roosevelt in January 1928, three years ago is relevant. At that time he said: "The unemployment is still employment; it builds no consumer goods for a living properly. The nation has the day when these weapons of destruction may be used against their neighbors or when an unusual emergency, like a house of cards, will fall."

It is far better to meet expansion needs with double shifts and thereby to reach an absolute limit upon expanding than to be required to expand four years to require but one shift during the air force expansion period and then have a semi-idle industry thereafter. To maintain our new air force after 1941, a complete month-shift industry is necessary as the basis of present estimates.

Let us not expand any further than that suggested, at least until there is a commercial need delivery in sight, as might well maintain should the very usual policy now covering some attention, of eliminating surpluses from all we build, becomes law in the interim. This policy, incidentally, is now in effect in Holland and Great Britain. An over-expanded industry would result not only in idle plant facilities but also in a vested government problem for maintaining military expansion.

(Continued on page 22)



THE EAGLE is CLEAR EYED

The eagle has long been noted for its keenness of vision. Its modern namesake, the "Eagle of Iron," is also distinguished in this respect, for, like all other Spartan executive models, this plane is equipped with crystal-clear Plexiglas windows and a curved Plexiglas windshield.

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Table 13—The Airplane and Engine "Dollars"

	Planes	Engines	
1938	100	100	100
1937	77	77	77
1936	55	55	55
1935	44	44	44
1934	33	33	33
1933	22	22	22
1932	11	11	11
1931	6	6	6
1930	3	3	3
1929	2	2	2
1928	1	1	1
Total	262	262	262

Notes: 1938

1938: 100% of 1938 is based on 100% of 1938. 1937: 100% of 1937 is based on 100% of 1937. 1936: 100% of 1936 is based on 100% of 1936. 1935: 100% of 1935 is based on 100% of 1935. 1934: 100% of 1934 is based on 100% of 1934. 1933: 100% of 1933 is based on 100% of 1933. 1932: 100% of 1932 is based on 100% of 1932. 1931: 100% of 1931 is based on 100% of 1931. 1930: 100% of 1930 is based on 100% of 1930. 1929: 100% of 1929 is based on 100% of 1929. 1928: 100% of 1928 is based on 100% of 1928. 1927: 100% of 1927 is based on 100% of 1927. 1926: 100% of 1926 is based on 100% of 1926. 1925: 100% of 1925 is based on 100% of 1925. 1924: 100% of 1924 is based on 100% of 1924. 1923: 100% of 1923 is based on 100% of 1923. 1922: 100% of 1922 is based on 100% of 1922. 1921: 100% of 1921 is based on 100% of 1921. 1920: 100% of 1920 is based on 100% of 1920. 1919: 100% of 1919 is based on 100% of 1919. 1918: 100% of 1918 is based on 100% of 1918. 1917: 100% of 1917 is based on 100% of 1917. 1916: 100% 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which would apprise a designer and understate size of efforts, both from the viewpoint of national economy and international order.

I believe that we can consider the present program involving an Army air force of six thousand planes and a Navy air force of three thousand as fundamentally sound. Consideration of rapid replacement delivery from the size of standing air force which we efficiently maintain. Twenty-four months for production delivery after a prototype has been developed is a long time in this rapidly advancing art.

We have initially not reached the stage of standardized design and are actually doing just what we wish, in size, shape, type of European nations. We do want and by our present program will have a sufficiently large air force to meet any threat to our interests, to provide freedom for training plans, and for preparing the industry.

A main thrust, supported by the future is a general emphasis on research facilities. Research, implying quality, remains the foundation of an efficient air force.

Our original assumptions definitely changed from analysis, consideration of emergency expansion which might be caused by a national emergency as major war. How to meet emergency expansion really forms the subject for another article, but a word or two here on the modern capacity in order. In the first place, the present industry geared to two shifts cannot be called emergency, even though emergency conditions would call for greater capacity. To meet that need, one possibility would be an increase in sub-contracting. This cannot be efficiently extended very much as regards the airplane although the engine and propeller business of the industry, particularly greater production of sub-contracting are present. These, undoubtedly, could be efficiently carried out in the automotive industry, perhaps on the basis of material tools manufactured by the aircraft engine and propeller companies. In any event, engineering control by the aviation industry, plus management advice is essential.

Much could be written on the "shadow" industry as worked out in England. Certainly, government financing for assembly plants is sound,

there to be used for emergency only. Here again, the experience at England has definitely demonstrated the advisability of maintenance of engineering control by the engine developing a given airplane or engine design.

A third type of expansion and one involving many desirable features might be the financing of extensions to existing plants of the industry with government funds, the supplementary facilities to be rented by the factory involved from the government, and remaining with the government after the emergency had passed. One thing of which we are certain is that expansion should not be means of government-owned and operated aircraft manufacturing plants. Experience here and in many countries already amply bears out this contention. Such plan is, perhaps, most definitely condemned by study of the situation which unfolded in France during the two or three years of the nationalization period. There were several months during the period when one company in this country delivered as many planes as France delivered as many companies in France which employed a personnel ten times as great! Aside from the probable failure of such factories in time of emergency, there would also be the matter of destructive postwar government competition.

There is the possibility of reducing the number of design and designing companies, setting up a group of production companies, number of whom could contract designs to production engineering from one development firm. This plan is, to a certain extent, followed in Germany and, from an efficiency standpoint, has much to commend it. Here again, however, need for engineering, follow up and control by the original designing firm is essential. For our country the scheme does not appear desirable.

Passing, as we do, adequate plans for air force expansion and facilities for carrying them out, let us now look down to the job of carrying them through to completion within the time limit set. Let us look to the plan and see if through without digressing too much. Let us also sound our research bellows, thus establishing a sound foundation for an efficient air force.

Such planned action, carried to conclusion, will do much to convince other countries of our ability, quickly and efficiently to take care of further emergency expansion should such unfortunate contingency be forced upon us.



"REQUIRED NO HEAT WHATSOEVER WITH NO LOSS OF POWER AT ALL"

We quote from a letter from Mr. W. A. Hamilton, System Maintenance Superintendent of Transcontinental & Western Air, Inc.

"Fleet Don B. Terry reports—when climbing up through a stratosphere foot overcast on November 16th of light air, where we were released over the top to Denver I could not be any other way than enthusiastic about the operation of the carburetors. The conditions were such that I feel sure there would have been considerable loss of power with the old type carburetor, necessitating the application of heat to maintain the necessary R. P. M., whereas these required no heat whatsoever with no loss of power at all. The rate of climb was excellent and after experiencing this performance I really feel you have an excellent carburetor for icing conditions and would not feel at all hesitant in flying in any weather we did in the DeS's."

Technical data or Aircraft Manual will be supplied on request.

HOLLEY CARBURETOR COMPANY
Aircraft Division Detroit, Michigan

NON-ICING
Aircraft
CARBURETORS

HOLLEY

AVIATION
June, 1941

Table 15—Cause of Increase in Price of Airplanes

	1939	1941	1940
1—Rise in Cost of Raw Materials	100	100	100
2—Rise in Cost of Labor	100	100	100
3—Increase in Material Costs	100	100	100
4—Increase in Production of Components	100	100	100
5—Increase in Transportation and Distribution	100	100	100
6—Increase in Overhead and Profit	100	100	100
7—Total Increase over 1939	100	100	100
8—Total Increase over 1939	100	100	100
9—Total Increase over 1939	100	100	100
10—Total Increase over 1939	100	100	100
11—Total Increase over 1939	100	100	100
12—Total Increase over 1939	100	100	100
13—Total Increase over 1939	100	100	100
14—Total Increase over 1939	100	100	100
15—Total Increase over 1939	100	100	100
16—Total Increase over 1939	100	100	100
17—Total Increase over 1939	100	100	100
18—Total Increase over 1939	100	100	100
19—Total Increase over 1939	100	100	100
20—Total Increase over 1939	100	100	100

Table 16—Comparison with Other Years and Other Countries

	1939	1940	1941
1—Total Increase over 1939	100	100	100
2—Total Increase over 1939	100	100	100
3—Total Increase over 1939	100	100	100
4—Total Increase over 1939	100	100	100
5—Total Increase over 1939	100	100	100
6—Total Increase over 1939	100	100	100
7—Total Increase over 1939	100	100	100
8—Total Increase over 1939	100	100	100
9—Total Increase over 1939	100	100	100
10—Total Increase over 1939	100	100	100
11—Total Increase over 1939	100	100	100
12—Total Increase over 1939	100	100	100
13—Total Increase over 1939	100	100	100
14—Total Increase over 1939	100	100	100
15—Total Increase over 1939	100	100	100
16—Total Increase over 1939	100	100	100
17—Total Increase over 1939	100	100	100
18—Total Increase over 1939	100	100	100
19—Total Increase over 1939	100	100	100
20—Total Increase over 1939	100	100	100

Note: 1) Data for 1939 (Pentagon) equals about 30% more than DeS.

2) Data for 1940 (Pentagon) equals about 30% more than DeS.

3) Data for 1941 (Pentagon) equals about 30% more than DeS.

4) Data for 1942 (Pentagon) equals about 30% more than DeS.

5) Data for 1943 (Pentagon) equals about 30% more than DeS.

6) Data for 1944 (Pentagon) equals about 30% more than DeS.

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134) Data for 2072 (Pentagon)

Stacks and Rings

(Continued from page 61)

read individually it is necessary to have a roomier panel with one more extension than the number of cylinders being read. This control panel is connected to the pump. By means of a short rubber tube jumper this gauge outlet can be placed in such of the other connections as there are a reading taken. An exhaust pump has been devised for this purpose.

It is necessary to check the back pressure in full throttle flight at 2000 revs stroke and also, if a supercharged engine is installed, at the critical altitude of the engine.

Carbon Monoxide (Intermittent)

The instrument necessary used for measuring the carbon monoxide content in the engine or intake air is manufactured by the Blue-Label Airplane Company. The operation of this instrument involves getting a steady flow of air under test through a hypodermic coil. The hypodermic coil acts as a catalyst in converting the carbon monoxide in the test sample into carbon dioxide. This conversion generates a quantity of heat which is directly proportional to the amount of carbon monoxide present in the sample. This heat is measured by a series of thermocouples which are connected to a millivoltmeter. This is the measuring meter. It is calibrated to read directly in hundredths of one percent. One instrument is available that is calibrated to read from 0 to 0.1% on one scale and from 0 to 1% on another scale. The first scale is more applicable to aircraft use since the permissible concentration from 900 to 0.1% for different altitudes. 0.1% carbon monoxide content is the maximum permissible concentration if mild poisoning and headaches are to be avoided.

The standard carbon monoxide indicator is operated by self-contained dry cell batteries so an installation consists of mounting it in the airplane in a location visible to the observer. Another model is designed to tap into the vacuum pump line to obtain a steady flow of the sample air. The indicator should be rubber mounted to prevent the sensitive self-contained dry cell batteries so an installation consists of mounting it in the airplane in a location visible to the observer. Another model is designed to tap into the vacuum pump line to obtain a steady flow of the sample air. The indicator should be rubber mounted to prevent the sensitive self-contained dry cell batteries so an installation consists of mounting it in the airplane in a location visible to the observer. Another model is designed to tap into the vacuum pump line to obtain a steady flow of the sample air. The indicator should be rubber mounted to prevent the sensitive self-contained dry cell batteries so an installation consists of mounting it in the airplane in a location visible to the observer.

pump in the indicator will suck the test sample through the tube and into the hypodermic coil.

Before reading check into the carbon monoxide indicator must be run for 10 minutes with the sampling tube and a new air. This is done on the ground before the engine is started when battery operation is used and means a thorough cleaning out of the indicator. The self-indicator is then adjusted to zero and the test begun. With full rich mixture the engine is flown between 1000 and 2000 full throttle at full throttle, and at cruising power. The sampling line is held at various points around the cockpit for a minimum of 2 minutes and a reading taken. After each reading is taken it is advisable to reset the sampling tube out of the engine for a short time to clear out the indicator.

Temperature Measurement

In the stored type of exhaust air probe the aluminum coating is set two inches away from the exhaust collector. If the stored unit is not vented the temperature would soon increase to a point where the coil would burn. When exhaust ducts are used the temperature within the exhaust must not exceed 600° F. If excessive and heat resistant steel sheet or ducted sheet is used for the stored the permissible temperature may be increased to 570° F. The temperature in the stored can be readily measured with the ordinary type thermometer.

Recommendations

The present trend in aircraft engine development is to increase the power by adding a second bank of cylinders thus keeping the diameter and frontal area the same. Therefore the exhaust power output by 50% to 100% will require up to a 50% increase in the diameter of the exhaust collector tubing. Locating their larger tubes within the available space will be very difficult and these are bound to be interference with other units of the engine. The enormous quantity of heat thrown off by the exhaust collector will be one of the main causes of trouble. The following recommendations are recommended to protect the important parts of the engine installation from this excessive heat.

1. Double Mufflers. The exhaust collector of high power engines are so close proximity to the intake manifold. There is a loss of 1% in power for about every 2° F increase in the temperature of the mixture in the intake manifold. This

loss in power is not registered on any instrument in a modern airplane but is nevertheless a loss of power that affects performance and should be checked. Wrapping the intake manifold with asbestos is one method of protecting them from the exhaust heat. Another suggested method is the use of a highly polished aluminum or 18-8 stainless tubing and shield the exhaust to the intake manifold and ducts to keep the exhaust heat.

2. Fuel System. At 20,000 feet altitude gasoline will boil at 190° F. It is difficult to keep an engine necessary components below this temperature with a large exhaust collector mounted just forward. It will be particularly important to keep the fuel feed in a stratoplane plane flying at 20,000-25,000 feet with supercharged high power engines. The development of a fuel meter or radiator is one solution.

Another method with which the author has had some success consists of a 12 inch flexible tube that leads from the front of the engine back to the fuel system. By this means a cooling jet of air is directed on the largest reservoir of fuel in the engine necessary component. The colder air last available at altitude compensates for the lower boiling point of gasoline due to reduction of pressure with altitude. In respect of the cooling method used it is advisable to run the fuel lines as direct as possible in the necessary components to minimize the absorption of heat.

There is a strong possibility that engines will be located within the wings, or mounted pusher type airplanes will be developed within the next few years. In these airplanes it will be practical to discharge the exhaust gases through slots in the tailing edge of the wing. By so doing a propeller drive will be obtained. In addition it is believed the flow over the engine wing will be improved. These two efforts will considerably increase the overall efficiency.

On all new airplanes it is recommended that the exhaust meter be located aft and behind the wing so the passing air will create a suction. By this means the back pressure within the collector can be reduced with a consequent gain in power.

Another improvement in performance is reducible by installing vent in the exhaust meter to smooth the flow of the exhaust gases. This is particularly true when the exhaust meter is near a wing.

(This is the fact of the story between Parts I and II appeared in Aviation for April and May, 1935)



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★The school has long in specifying terms years and was formerly open only to boys who qualified also extremely high academic level and who maintained rigid academic standards. NOW the school is open to boys of all levels, of high mental caliber who wish to work hard to qualify for a degree before the highly specialized field. Only boys who are graduates of approved high schools and whose academic standing is in the top quartile of their classes will be accepted as students. They should, preferably, have major in mathematics and the sciences but may take this studies in their first year with the freshmen and transfer to college subjects. Students enter in June or September and graduate in September or June two years later. The course runs through the summer months with four weeks during each 12 month period for vacation. The school has been opened to make this course the finest available and is setting up a school of aviation type for the specialized education.



A student working on a model airplane in a workshop. The school has a large shop for building and repairing aircraft. The school also has a large library and a large dining hall.



A student working on a model airplane in a workshop. The school has a large shop for building and repairing aircraft. The school also has a large library and a large dining hall.

The basic course is commercial engineering for which an average of 4000 man hours is required for the first year. This is approximately 700 hours more than an ordinary four-year engineering college degree. A flight course of 300 hours is optional and includes advanced blood flight and other procedures as a part. In modern flying equipment ranging from 30 to 250 h.p. Students may take the commercial engineering course to be eligible for flight training. Aeronautical Engineering leads to a Bachelor of Science degree in Aeronautical Engineering. Flight students receive commercial licenses from the Civil Aeronautics Authority and all students receive engine and engine mechanics licenses during their instruction period. When you graduate from Lewis Holy Name School of Aeronautics you are fully equipped to enter the Aviation field and move up to the qualifications demanded by the aircraft position. Our well equipped machine and airplane school by these give students a thorough grasp that means which is an absolute essential in a complete aeronautical education. The school is the right to give you more practical facts.



TO PARENTS:

If you are thinking about it take a few minutes and carefully examine and discuss it with your family. We have a large library and a large dining hall. We have a large library and a large dining hall. We have a large library and a large dining hall.

STUDENT'S FOR INFORMATION: We are looking for boys who are interested in aviation and who are capable of high academic achievement.

FACTS ABOUT THE SCHOOL

- **OPERATION**—The school is under the personal supervision of the Rev. Monsignor John J. Lewis, S.J., who is also the principal. The school is located in Lockport, Illinois, and is open to boys of all ages. The school has a large library and a large dining hall.
- **EDUCATION**—The school is open to boys of all ages. The school has a large library and a large dining hall.
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In recent weeks 18 experienced pilots have received the instrument flight training, a procedure having come upon the explicit recommendation of chief pilots of major commercial air transport companies. At the time of writing, four have accepted or hope to accept positions.

Facts

Facts about
PARKS AND COLLEGE

[illegible]

PARKS AIR COLLEGE
EAST ST. LOUIS, ILLINOIS

O. K.
FOR PRECISION
WORK



Fig. 1. (A) Hypodermis
epithelium. (B) Hypodermis
epithelium. (C) Hypodermis
epithelium.

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A few of the styles available



Fig. 101.—"Kathwell" pose in relationship with legless snake in space. See the Index and Marked, etc.



Fig. 125—Hollander, Iowa.
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top of hickory wood. See
the table for the work.

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Recent analysis are indispensable in long-range aircraft where maximum fuel economy is imperative. Special types are available for unusual installations.

(Top) Histogram, non-linear analysis. Small histogram, automatic dtd, single region case.



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WHEN SODIUM LIVES IN A GLASS VALVE

Look closely at the glass valve in the left-hand picture and you will see the sodium inside. It is a molten metal, and it is the reason the valve works. The right-hand picture shows the same valve, but the sodium has been removed. The valve is now a hollow shell, and it will not work.

• The molten sodium-filled valve has contributed much to the performance and dependability of today's aircraft engines. Because, which has the advantage of being between liquid inside the valve at 200° F. In the working valve it replaces the piston and by movement delivers the heat from the valve head to the water guide.

In the Thompson Products research department glass valves filled with sodium are used to observe the behavior of valves at high and low speeds, in the action of the liquid sodium may be observed under actual working conditions. These observations give some of the data for improved valve design and performance.

Thompson Products manufactures over two hundred different parts for aircraft engine and airplane builders. The sodium that lives in glass valves is typical of the company's manufacturing that keeps these many parts ahead of our nation's progress.

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These new generators produce 100 amperes at 15 volts and 50 amperes at 30 volts respectively with the

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